

# ***Measured solar spectral irradiance variability using the SORCE SIM***

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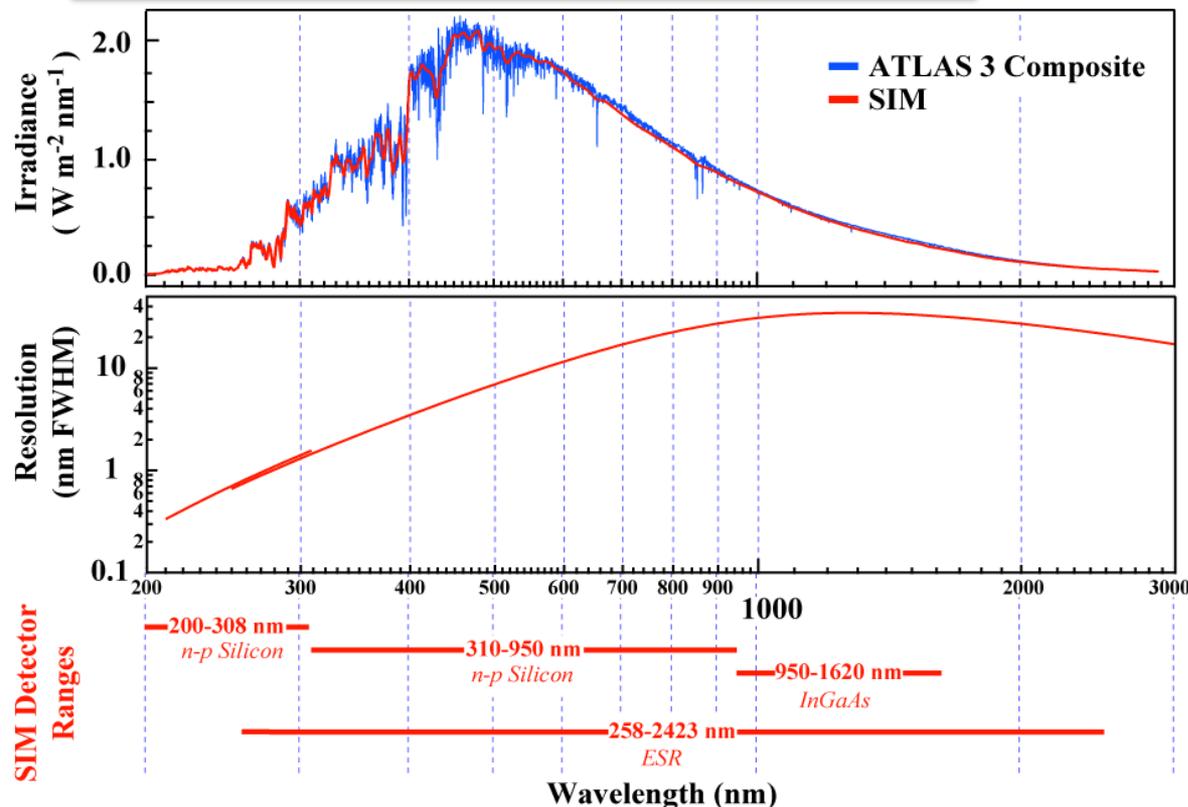
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***University of Colorado***

- **Review of the SIM instrument**
  - Detectors and spectral ranges, resolution, sampling
  - Instrument configuration
- **The SIM measurement equation**
- **Sources of noise**
  - ADC noise limited performance on photodiodes
  - Noise spectrum of the ESR
- **Sources of irreversible time dependent degradation**
  - Prism degradation – the dominant source of instrument sensitivity loss
  - Effects of spacecraft safe-hold events

# Instrument Overview

- **Instrument Type:** Féry Prism Spectrometer
- **Wavelength Range:** 200-2400 nm
- **Wavelength Resolution:** 0.24-34 nm
- **Detector:** ESR, n-p silicon, InGaAs
- **Absolute Accuracy:** 2-8%
- **Relative Accuracy:** ~0.5-0.02% (210-2400 nm)
- **Long-term Accuracy:** 0.3-0.02%/yr (210-2400 nm)
- **Field of View:** 1.5x2.5° total
- **Pointing Accuracy/Knowledge:** 0.016°/0.008°
- **Mass:** 21.9kg
- **Dimensions:** 88 x 40 x 19 cm
- **Orbit Average Power:** 17.5 W
- **Orbit Average Data Rate:** 1.5 kbits/s
- **Redundancy:** 2 Redundant Channels



$$\int_{\lambda=201}^{\lambda=2423} E(\lambda) d\lambda \approx 1324.49 \text{ Wm}^{-2}$$

≈ 97.3% of TSI

⇔ 36.32 Wm<sup>-2</sup> missing from TSI

Harder J. W., G. Thuillier, E.C. Richard, S.W. Brown, K.R. Lykke, M. Snow, W.E. McClintock, J.M. Fontenla, T.N. Woods, P. Pilewskie, 'The SOLAR SIM Solar Spectrum: Comparison with Recent Observations', *Solar Physics*, **263**, Issue 1 (2010), pp 3, doi:10.1007/s11207-010-9555-y

Paganan, J., J. W. Harder, M. Weber, L. E. Floyd, and J. P. Burrows, 'Intercomparison of SCIAMACHY and SIM vis-IR irradiance over several solar rotational timescales', *A&A*, 528, A67 (2011), doi: 10.1051/0004-6361/201015632, 2011.

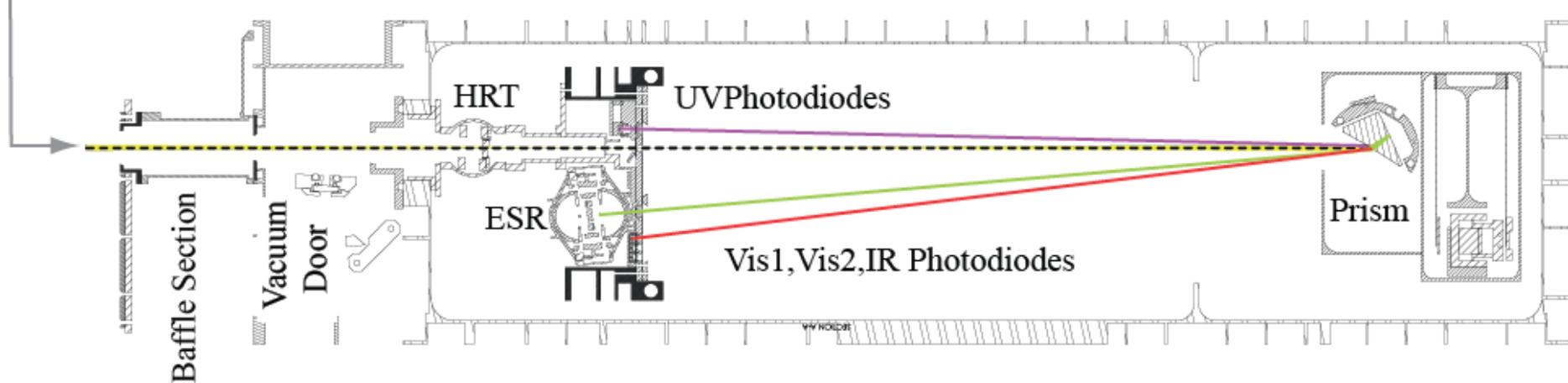
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- No assumptions are made about magnitude, slope, or time dependent behavior of SSI.
- Method of degradation correction is similar to the method used to correct TSI instruments but done as a function of wavelength.
- Degradation Corrections:
  - Exposure-related prism transmission loss accounts for the majority of sensitivity loss in SIM
    - Correct by comparing two spectrometers at different exposure rates
    - These spectrometers are in the same, physical, chemical, and thermal environments, so instrument changes are common mode.
  - Non-exposure related effects from space and spacecraft environment must be handled independently from prism transmission.
    - Correct ESR gain changes through routine gain measurement experiments
    - Correct photodiode detectors by comparing with the ESR's
    - Identify time periods when the instrument is affected by spacecraft disturbances

# Instrument Configuration

SIM A: Cross section in dispersion direction (SIM B mirror image)

$1/2^0$  Solar Input Beam



- **Stability of the ESR anchors the corrections for SIM**
  - Energetic photons do not make it to the ESR
  - Input flux is very small ( $<60 \mu\text{W}$ )
  - Critical power replacement resistors and 7.1 V reference are radiation hard
  - Weekly ESR gain measurements test and correct the ‘softer’ ESR electronics
- **Only one optical element to degrade through transmission loss**
- **Transmission loss probably arises from exposure of the prism to energetic solar photons:**
  - Directly induce compositional changes in the first few monolayers of the glass
  - transmission losses due to polymerization of trace amounts of organic materials on the surface

# Fundamental SIM Measurement Equation

$$E(\lambda) = \frac{\text{measuredDetectorCounts}(\lambda)}{\int \text{entranceApertureArea} \times \text{detectorResponseFunction}(\lambda) \times \text{spectralBandwidth}(\lambda) \times \text{OpticalTransmission}(\lambda) d\lambda} \quad (\text{units of } Wm^{-2}nm^{-1})$$

**ESR Power**  
(phase sensitive detection)

$$E_{ESR}(\lambda_s, t) = \frac{\frac{1}{M} \frac{V_7^2 R_h}{(R_h + R_s)^2} \left\{ \begin{array}{l} 1 + \tilde{G}(t) \vec{p} \cdot \vec{D} \\ \tilde{G}(t) \vec{p} \cdot \vec{Q} \end{array} \right\}}{A_{slit}(T) \int \alpha(\lambda) Tr_0(\lambda) \Phi(\lambda) S(\lambda_s, \lambda) d\lambda} \frac{1}{\left( (1 - a_{ESR}) \exp(-\kappa(\lambda)C(t)) + (a_{ESR}) \exp\left(-\frac{\kappa(\lambda)C(t)}{2}\right) \right)} \frac{1}{f_{1au}} \frac{1}{f_{doppler}}$$

**Orbit Correction**

**Detector photocurrent**

$$E_{Diode}(\lambda_s, t) = \frac{\frac{V_{max}}{M} \left\{ \frac{D - D_0}{R_f} \right\}}{A_{slit} \int \underbrace{Rs(\lambda, t, T)}_{\text{Profile Integral}} Tr_0(\lambda) \Phi(\lambda) S(\lambda_s, \lambda) d\lambda} \frac{1}{\left( (1 - a_{diode}) \exp(-\kappa(\lambda)C(t)) + (a_{diode}) \exp\left(-\frac{\kappa(\lambda)C(t)}{2}\right) \right)} \frac{1}{f_{1au}} \frac{1}{f_{doppler}}$$

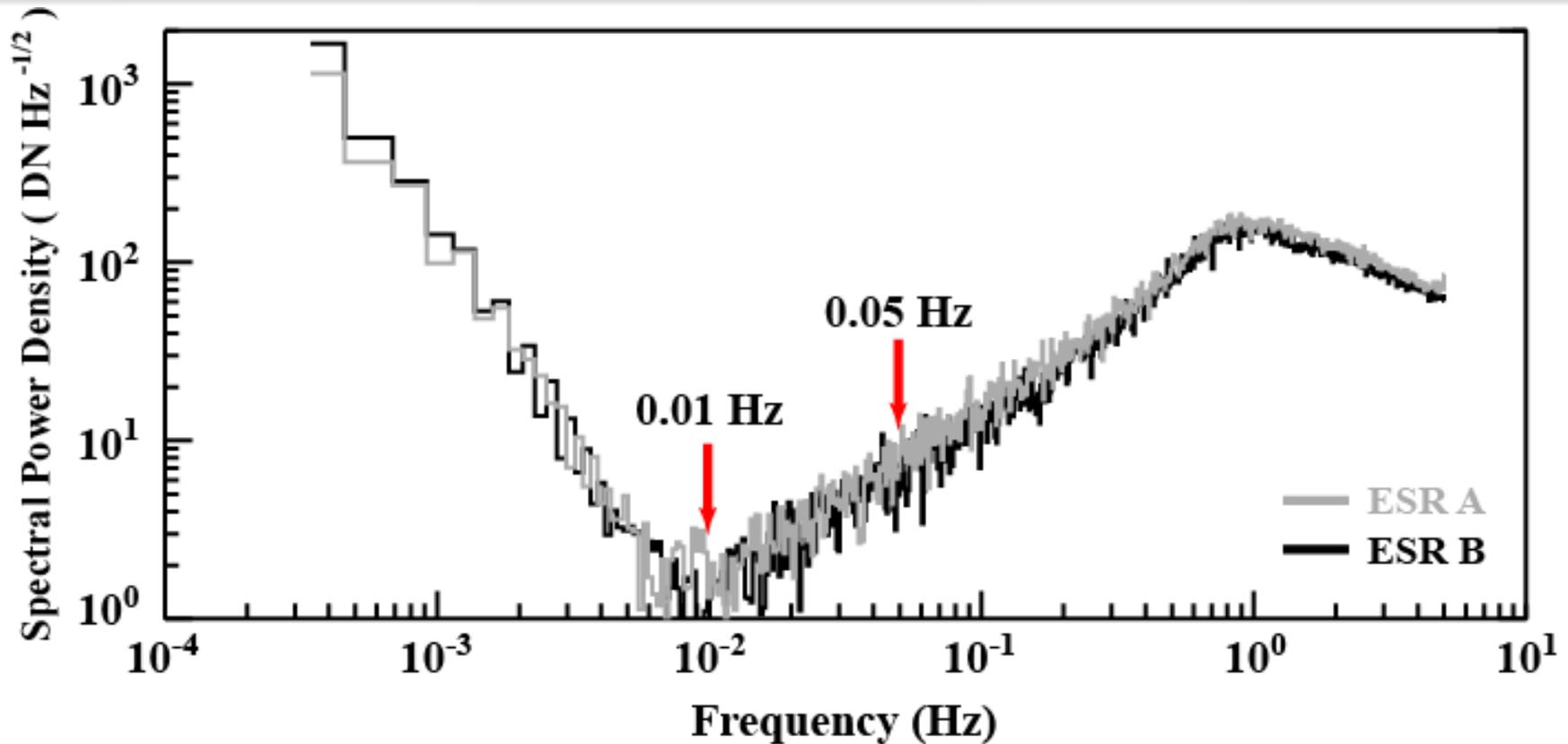
**Prism Transmission Degradation Factor**

- Exposure related degradation
- Non-exposure related degradation

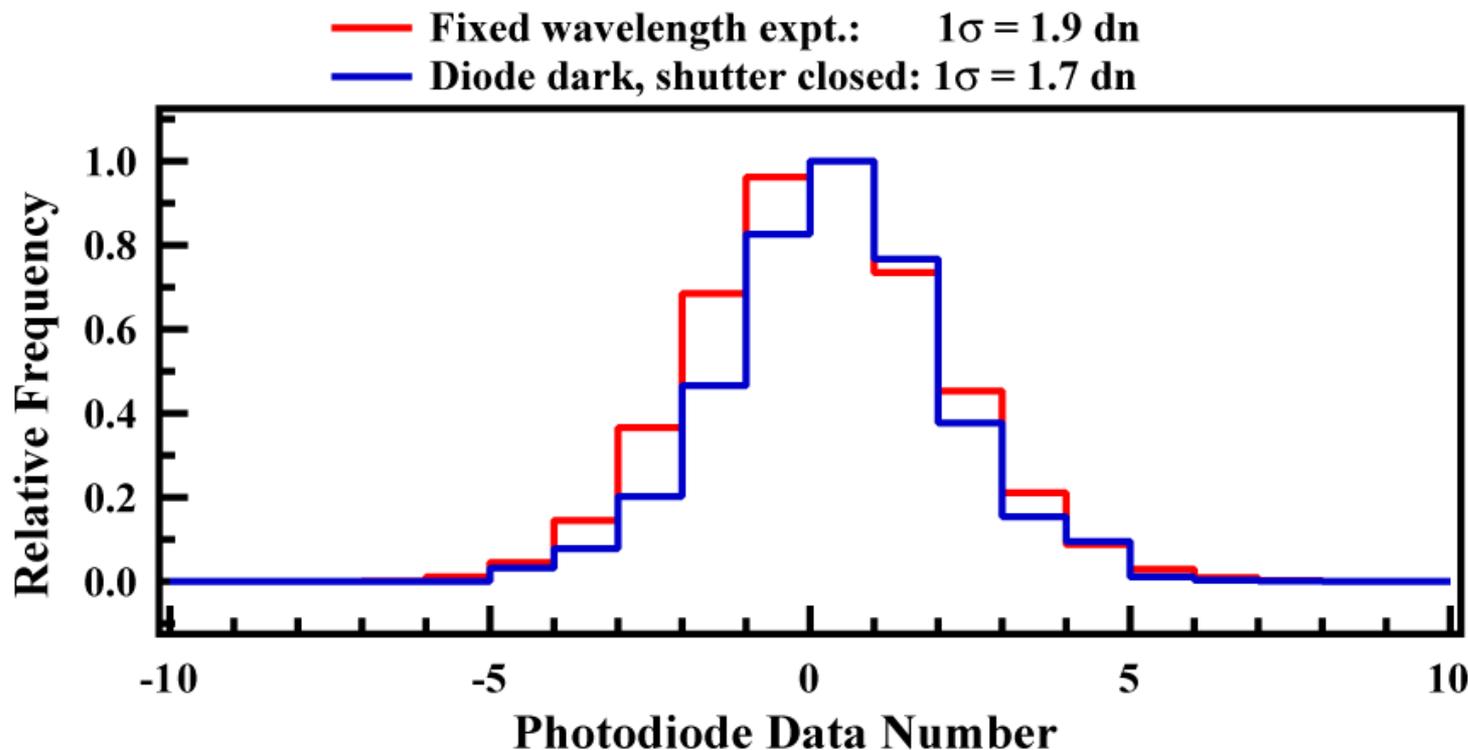
# Sources of Uncertainty in the SIM Time Series

Rank	Cause	Effect	Mitigation/magnitude of effect
<b>Short-term effects – do not accumulate with time</b>			
1	Detector Noise	Ultimate limit of comparison of two spectra	ESR : $10^{-3}$ to $10^{-5}$ Diodes; $3 \times 10^{-3}$ to $5 \times 10^{-5}$
2	Spacecraft pointing	Local perturbation in prism transmission/wavelength shift	Can produce spurious noise on the order of 1%, problem fixed through data masking
3	Detector Temperature	Spurious structure to photodiode data (700-900 nm range)	Adds about 500 ppm of noise at these wavelengths, refinements needed in processing.
4	Prism Temperature	Wavelength shift	Corrected to ~150 ppm in data processing
5	Scattered light	Increases apparent irradiance – decreases contrast in ‘lines’	<100 ppm in ESR, VIS, IR detectors, <0.5% in UV photodiode, not corrected
<b>Long-term effects – accumulate with time</b>			
1	Prism Transmission	Long-term reduction in instrument response.	Residual uncertainty 0.3-0.01%/yr
2	Optical alignment changes	Produces ‘jumps’ in the data at well-defined times	Problem significant after 2009/01/01 uncertainty ~0.1% in certain wavelength bands
3	Photodiode Radiant sensitivity	Reduction in photodiode sensitivity (750-950 nm range)	Comparisons to ESR; Comparable to diode noise $\sim 10^{-4}$
4	ESR servo gain degradation	Reduction in responsivity of the ESR detector	7 ppm/year, uncertainty ~1 ppm/year

# Detector Noise

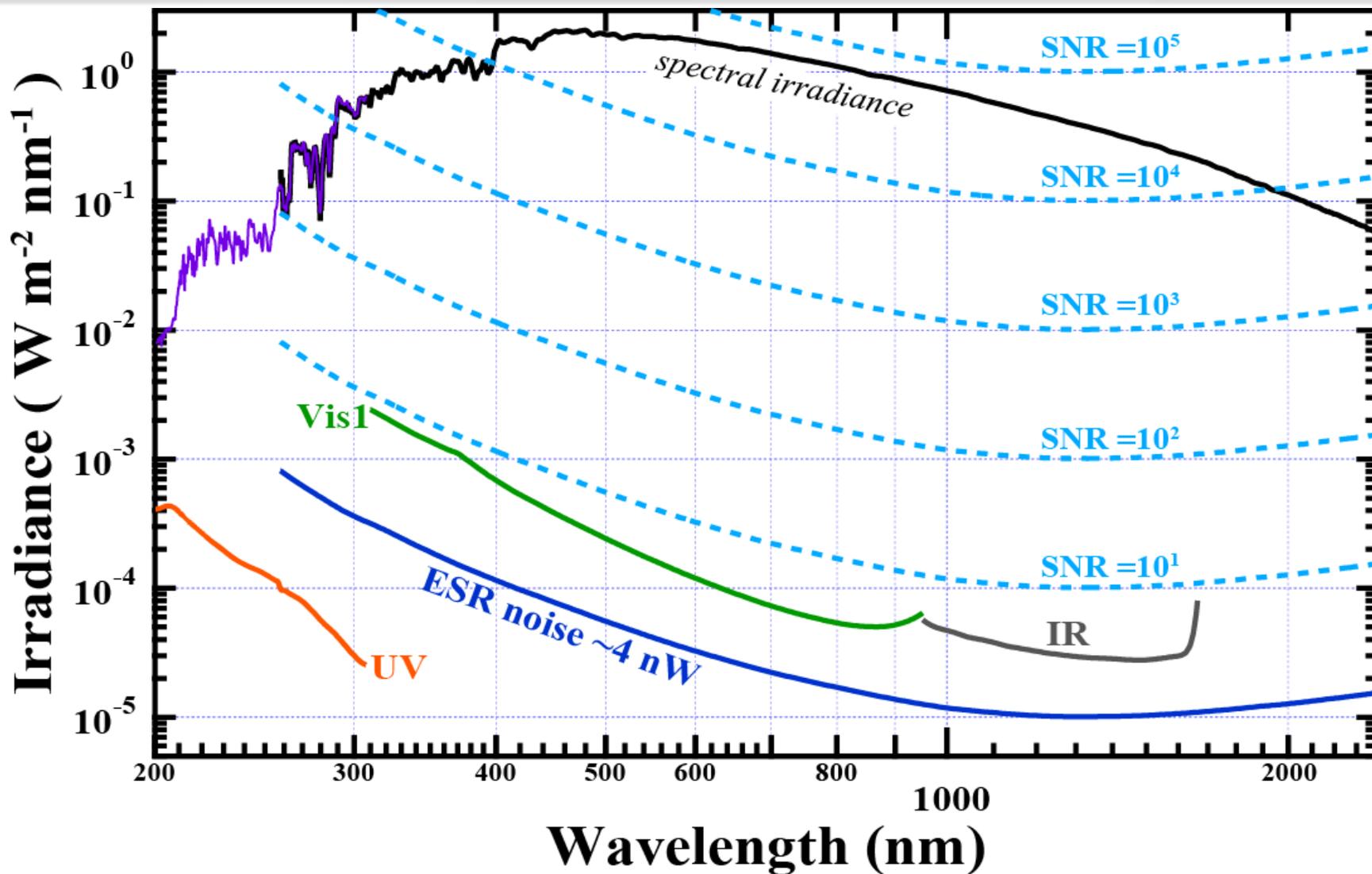


- Noise spectrum invariant with time
- Other effects ( $\lambda$ -shift, roll effects) can reduce the effective precision of the measurement
  - Typical noise level  $\sim 4$  nW, 50 sec half cycle, 2 cycles



- Photodiode noise determined by noise on the ADC - not photon noise.
  - Output of photodiode detectors multiplex and read-out by the same ADC, therefore noise level is common to all photodiode detectors.
- Noise levels have not changed over the course of the mission, independent of where in the orbit the spectrum is taken (i.e. no SAA effects)
- Distribution of noise independent of whether shutter is opened or closed.
  - Determined from daily darks and 2005/2006 fixed wavelength experiments

# Noise Equivalent Irradiance



Contours of signal-to-noise ratio are relative to the ESR 50 sec half-cycle, 200 sec integration

# Prism Degradation

- **Degradation is proportional to exposure**
  - SIM B has only  $\sim 1/4$  of the exposure of SIM A
  - A function  $F(t)$  can be found that produces the same trends in A&B channels
    - This is reasonable since the two instruments are in the same physical enclosure and their environment cannot evolve independently

# Determination of the Degradation Function

Lambert's Law:  $I(\lambda) = E(\lambda)e^{-\tau}$

For a single wavelength,  $\lambda$ . At two times 0 and 1.

$$\ln(I_{A0}) = \ln(E_0) - \tau_{A0}, \quad \ln(I_{A1}) = \ln(E_1) - \tau_{A1}$$

$$\ln(I_{B0}) = \ln(E_0) - \tau_{B0}, \quad \ln(I_{B1}) = \ln(E_1) - \tau_{B1}$$

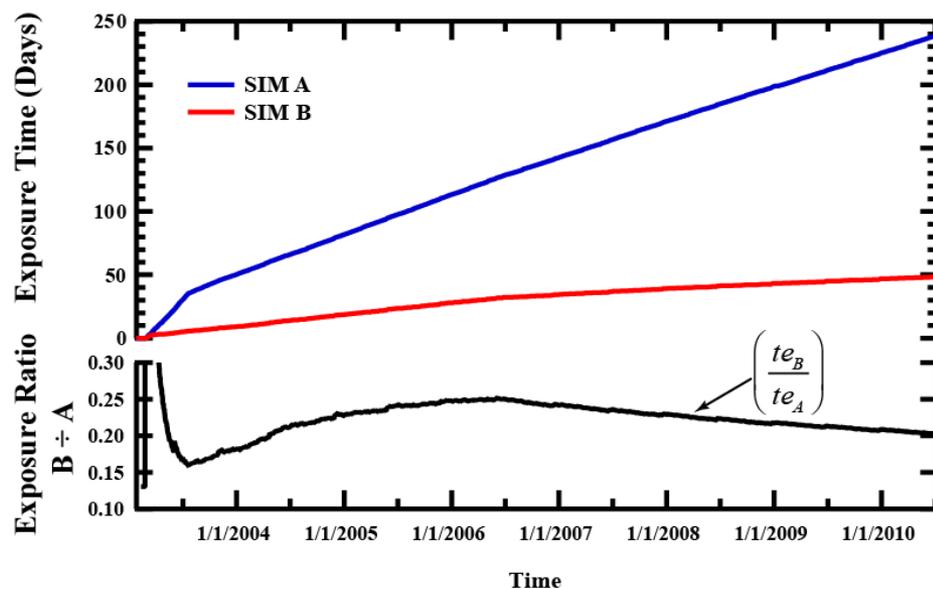
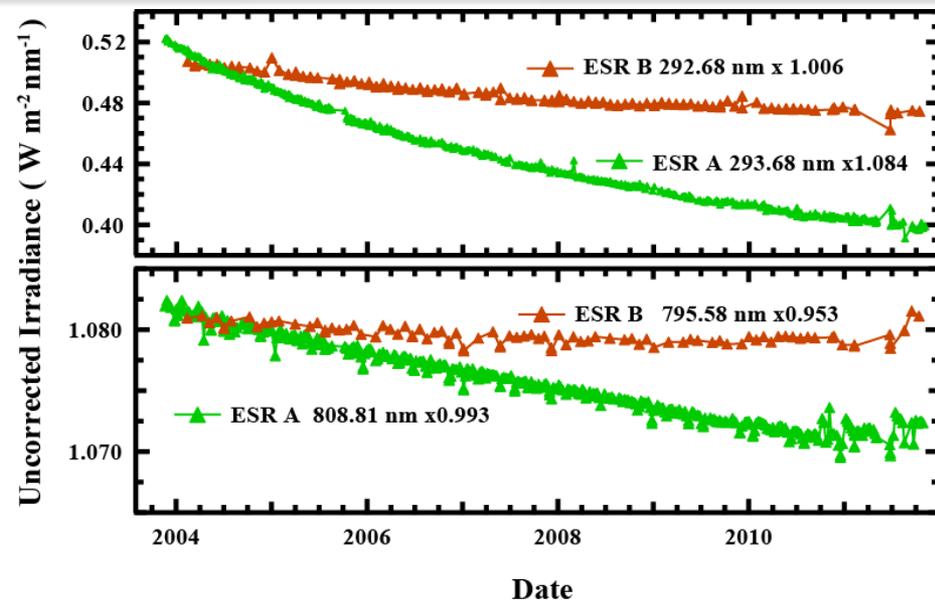
Combine and recase equations for SIM A & B in terms of measured quantities:

$$\text{For SIM A: } \ln\left(\frac{I_{A1}}{I_{A0}}\right) = \ln\left(\frac{E_1}{E_0}\right) - F(t) \cdot \Delta X_{A0 \rightarrow 1}$$

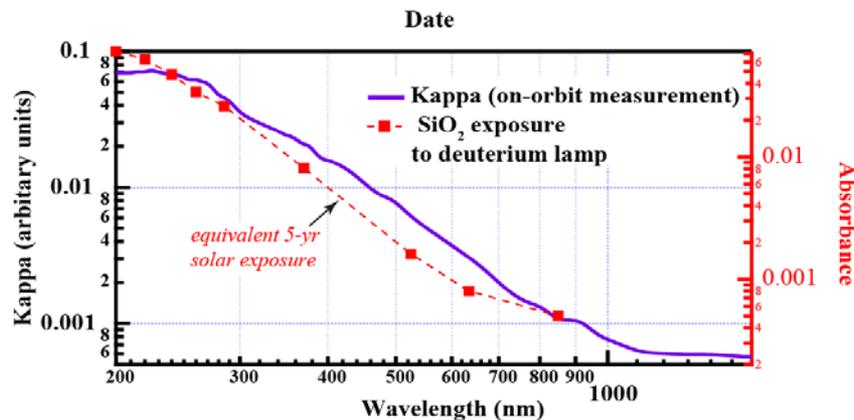
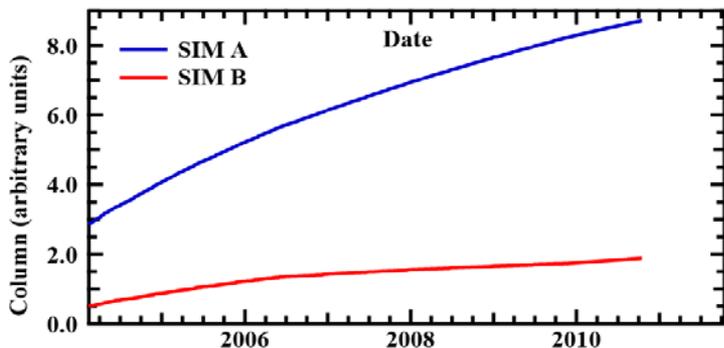
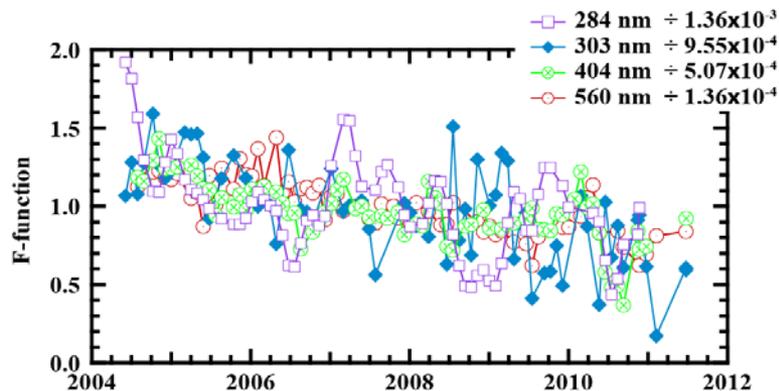
$$\text{For SIM B: } \ln\left(\frac{I_{B1}}{I_{B0}}\right) = \ln\left(\frac{E_1}{E_0}\right) - F(t) \cdot \Delta X_{B0 \rightarrow 1}$$

$F(t)$  = Time dependent degradation factor

$\Delta X_{A0 \rightarrow 1}$   
or  
 $\Delta X_{B0 \rightarrow 1}$  = Measured exposure time between times  $t=0$  and  $t=1$



# Determination of the Degradation Function



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$$F = \frac{\ln\left(\frac{I_{B1}}{I_{B0}}\right) - \ln\left(\frac{I_{A1}}{I_{A0}}\right)}{\Delta X_{A_0 \rightarrow A_1} - \Delta X_{B_0 \rightarrow B_1}}$$

Degradation accumulates with time and must be determined for each wavelength.

$$\tau_A(\lambda, t_1 - t_0) = \int_{t_0}^{t_1} F(\lambda, t') \frac{\partial X_{A_0 \rightarrow A_1}}{\partial t'} dt + Const.$$

(analogous for SIM B)

Decompose  $\tau(\lambda, t_1 - t_0)$  into two components:

$C(t_1 - t_0)$  a time dependent part

$\kappa(\lambda)$  a wavelength dependent part

Then:

$$\tau_{AorB}(\lambda, t_1 - t_0) = \kappa(\lambda) C_{AorB}(t_1 - t_0)$$

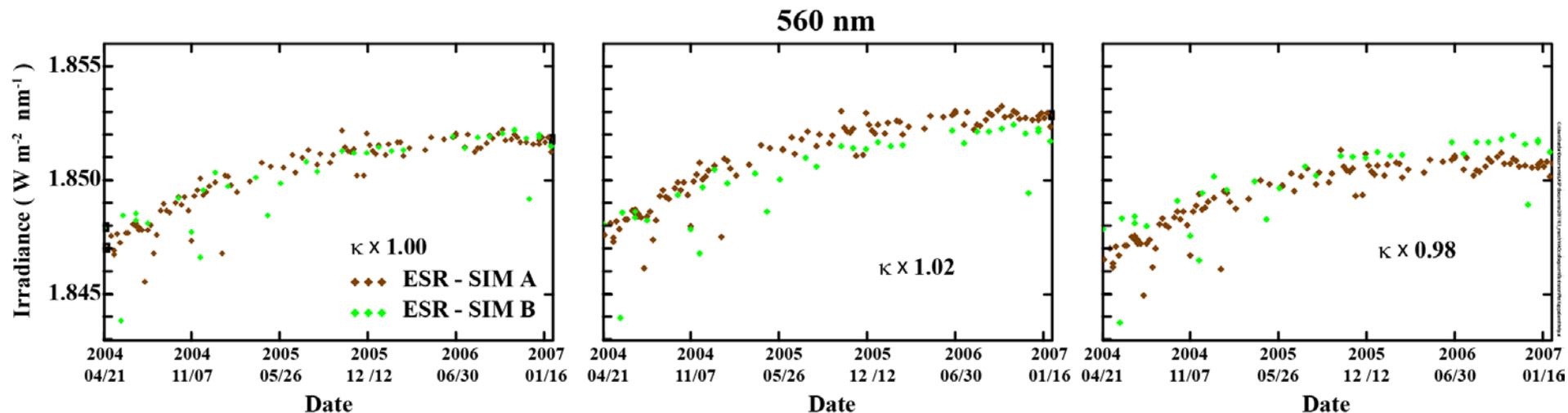
The degradation correction is:

$$E_{AorB}(\lambda, t) = \frac{I_{AorB}(\lambda, t)}{\exp(-\kappa(\lambda) C_{AorB}(t_1 - t_0))}$$

# Time Series of Degradation Corrected Irradiance

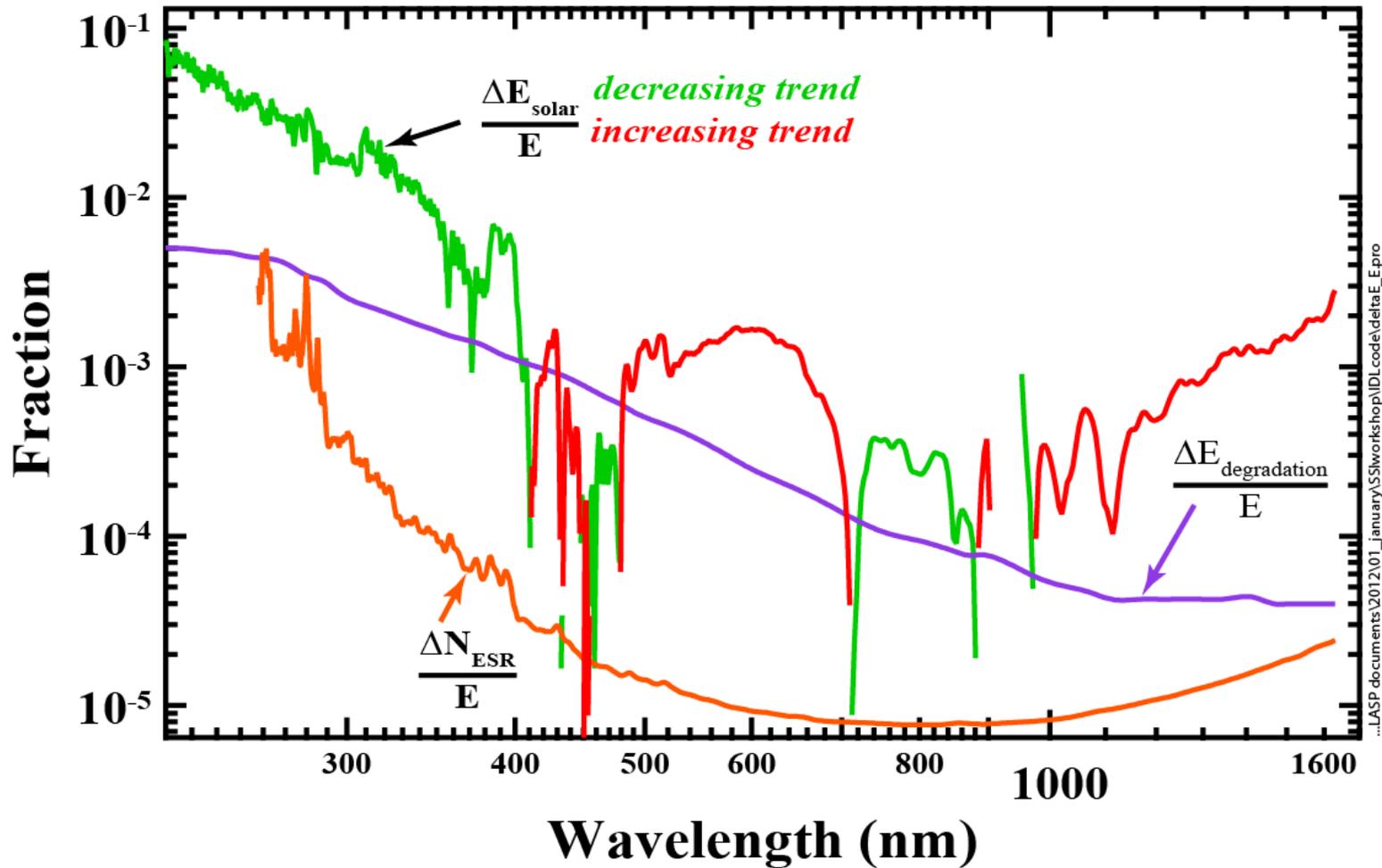
- Perturb the value of  $\kappa(\lambda)$ , apply to both SIM A & B
- Time series for the two instruments will diverge because the rates of exposure are different
  - Provides a method to estimate the uncertainty in  $\kappa(\lambda)$

$$\frac{\Delta \tau}{\tau} = \frac{\Delta \kappa}{\kappa} \quad \frac{\Delta \kappa}{\kappa} \leq 0.02 \quad (\pm 2\sigma)$$



$$\frac{\Delta E_{\text{degradation}}}{E} = \Delta \tau = \kappa(\lambda) (C_{t_1} - C_{t_0}) \frac{\Delta \kappa}{\kappa}$$

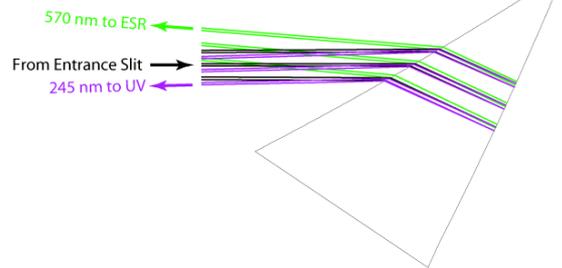
# Contributions of Uncertainty to trend



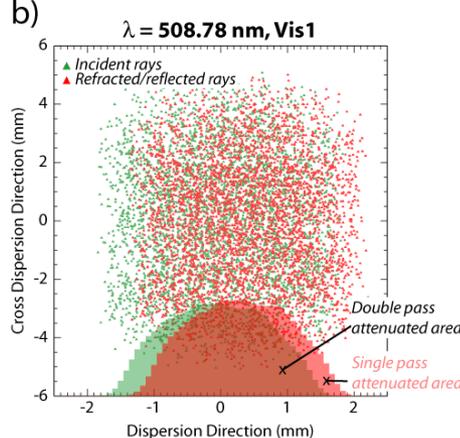
- Time range 2004/06/1-2007/11/15, 1262 days, 3.45 years
- $\Delta N_{esr} = 4\text{nW}$

# Transfer ESR correction onto adjacent photodiodes

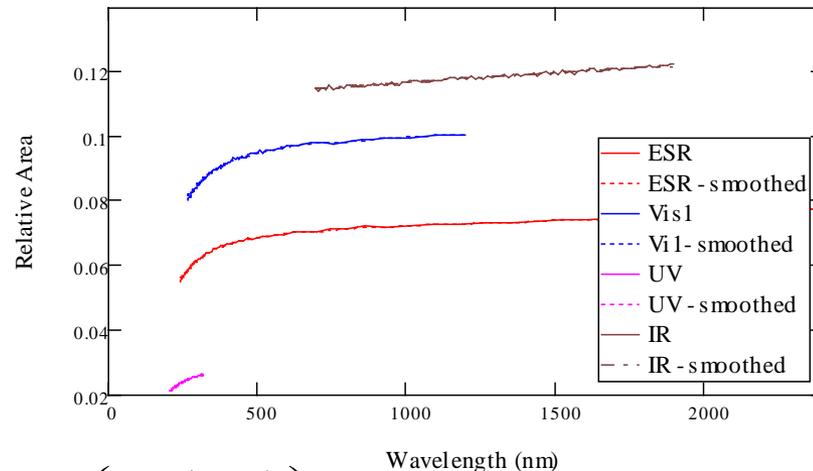
a)



b)



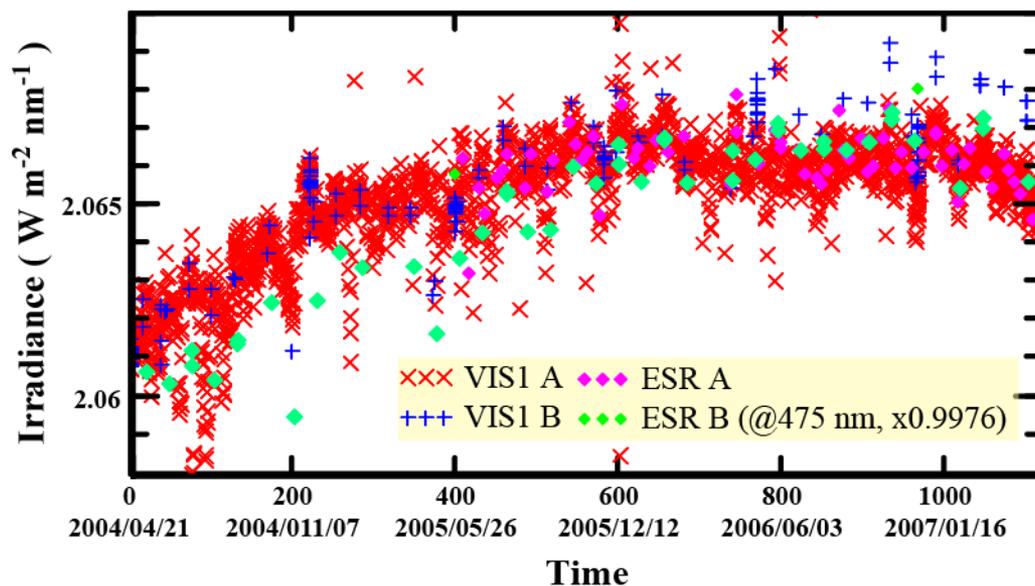
Single Pass Relative Area



$$\frac{E_{corrected}}{E_{uncorrected}} = (1 - a_d) \exp(-\tau(\lambda, t)) + (a_d) \exp\left(\frac{-\tau(\lambda, t)}{2}\right)$$

- No pristine area on the surface of the prism and different light paths through the prism encounter different amounts of absorbing material.
- $a_d$  initially estimated from ray traces and then adjusted to match the trends seen in the ESR

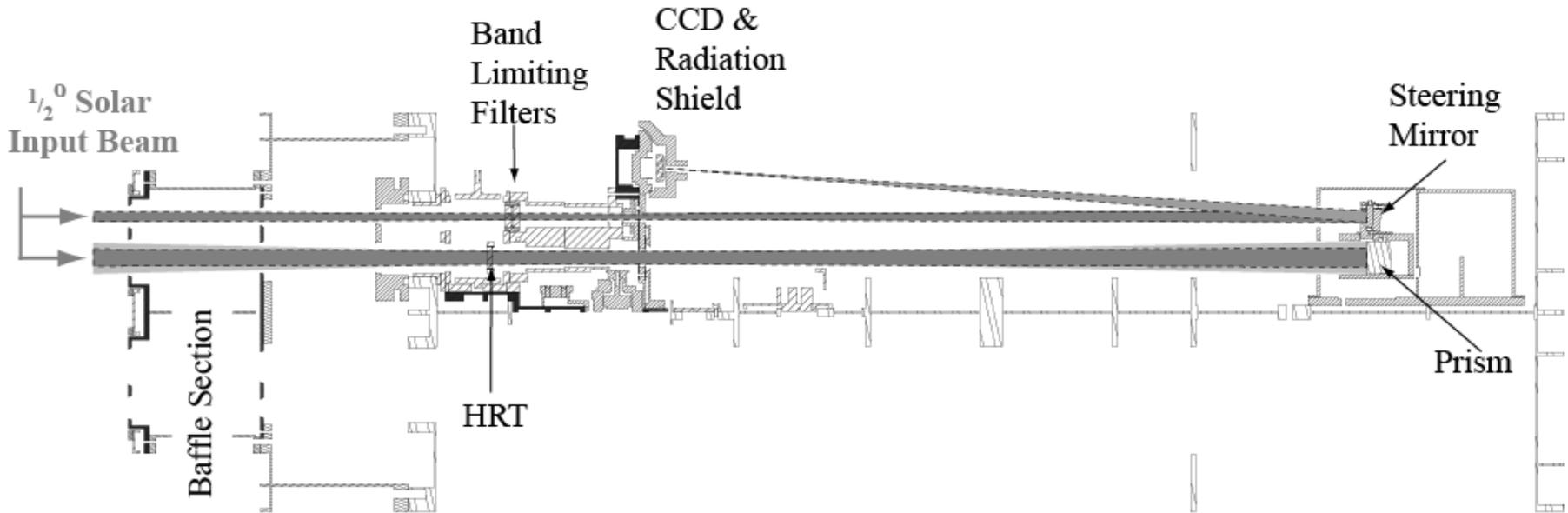
Degradation Corrected Irradiance at 472 nm



# Effects of OBC Anomalies and Wavelength Stability

# SIM in Cross Dispersion Direction

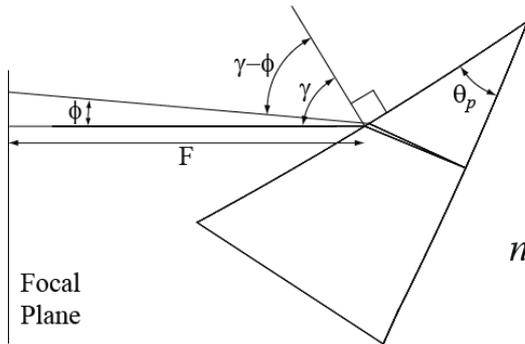
## SIM A: Cross Dispersion Direction



Date

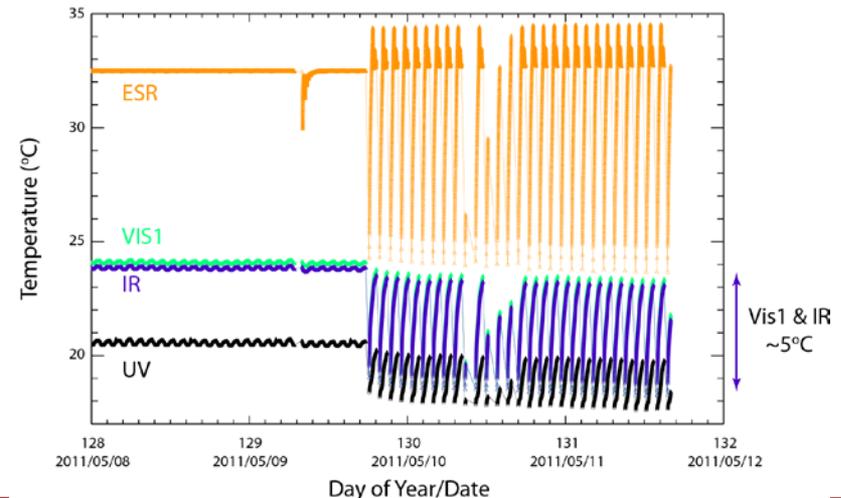
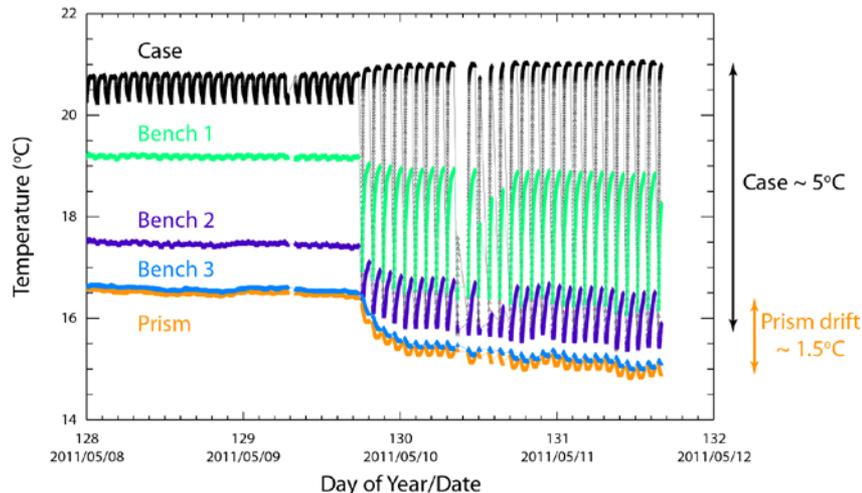
$$\gamma = (1 + \alpha) \gamma_z + \frac{1}{2} \tan^{-1} \left( \frac{(C - C_z) \times px(1 + \beta)}{F_{REF}} \right)$$

$$n_{@vis1} = \frac{1}{\sin(2\theta_P)} \sqrt{\sin^2(\gamma) + 2 \cos(2\theta_P) \sin(\gamma) \sin(\gamma - \phi) + \sin^2(\gamma - \phi)}$$

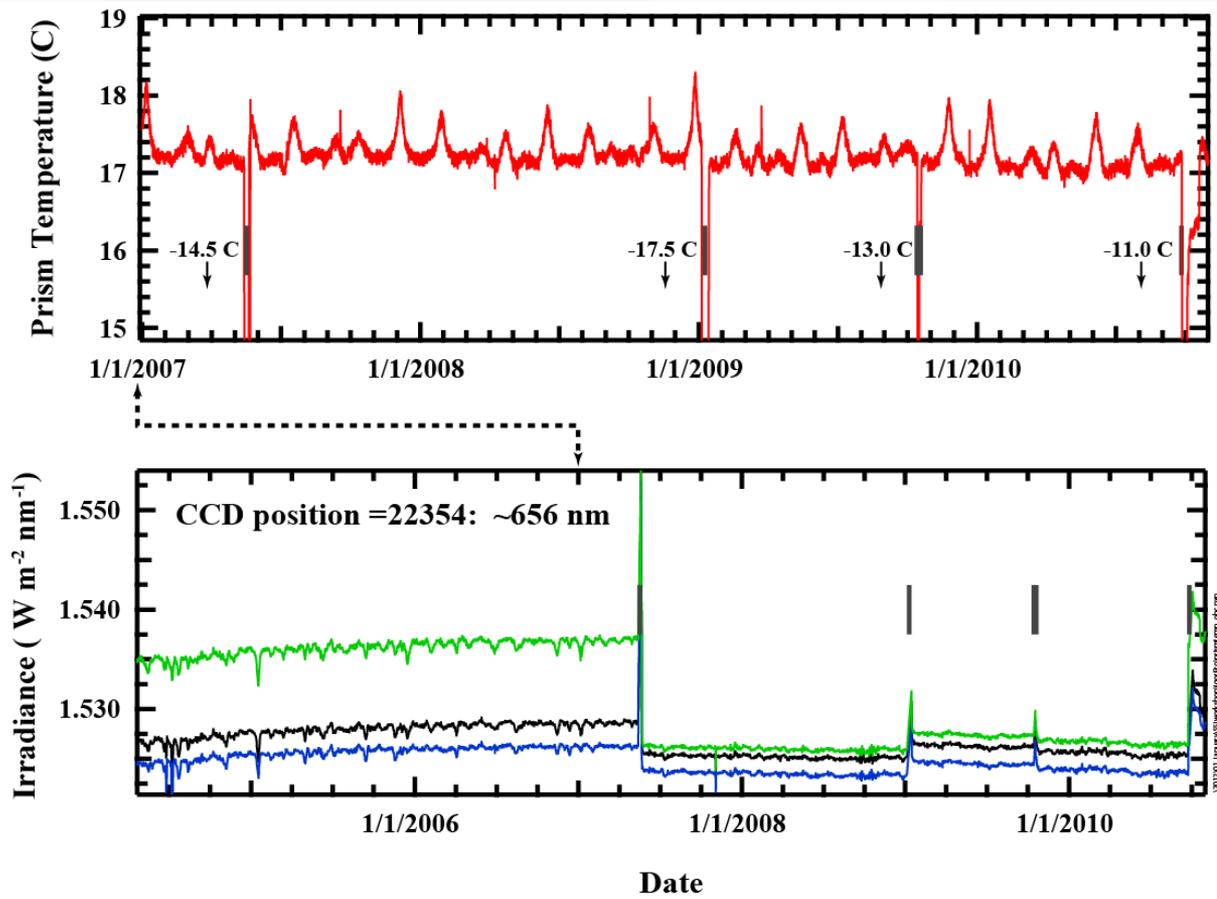


# Status of SIM after October 2010

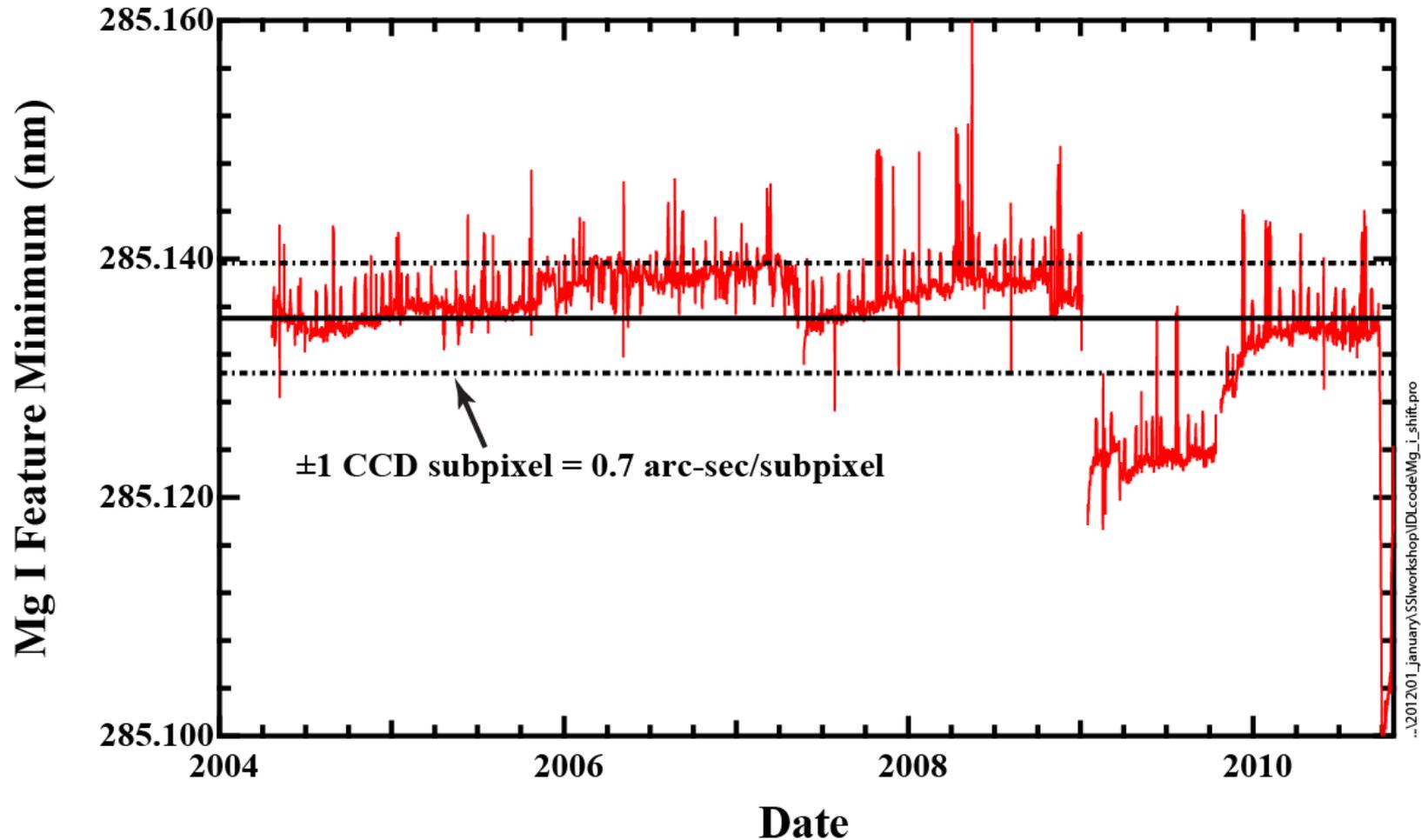
- 4 different safe-hold events since September 26, 2010 caused observatory temperatures to drop below  $-15^{\circ}\text{C}$  producing small but noticeable changes in the responsivity of SIM.
  - The cause of these anomalies has been corrected in flight software
- Because of decreased battery capacity, power cycling on every orbit started on 2011/05/04
  - Power cycling has no instrument safety issues and has not disrupted normal data acquisition
  - Detectors have 3-5 degree temperature swings
  - Prism temperature drifts continuously ( $\sim 1.5^{\circ}\text{C}$ ) but does not effect data processing
  - $\sim 1$  year of data may be needed to detect second order effects



Spacecraft Events
05/18/2007
01/09/2009
10/19/2009
10/14/2009
09/26/2010
12/26/2010
01/28/2011
05/13/2011
09/06/2011

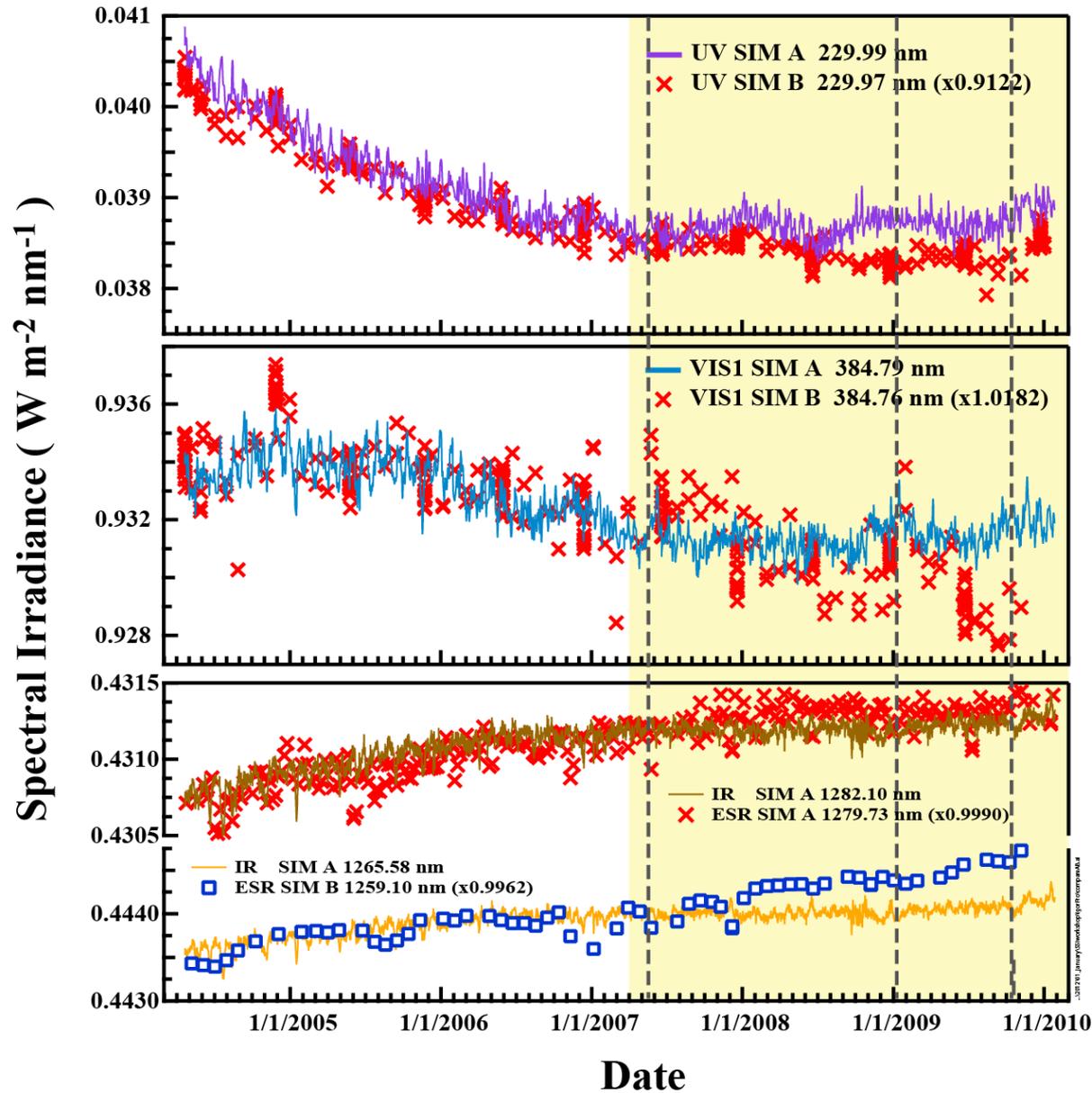


- OBC events cause a change in the orientation of the steering mirror relative to prism face, wavelength correction performed but it re-maps the CCD pixels onto the wavelength scale

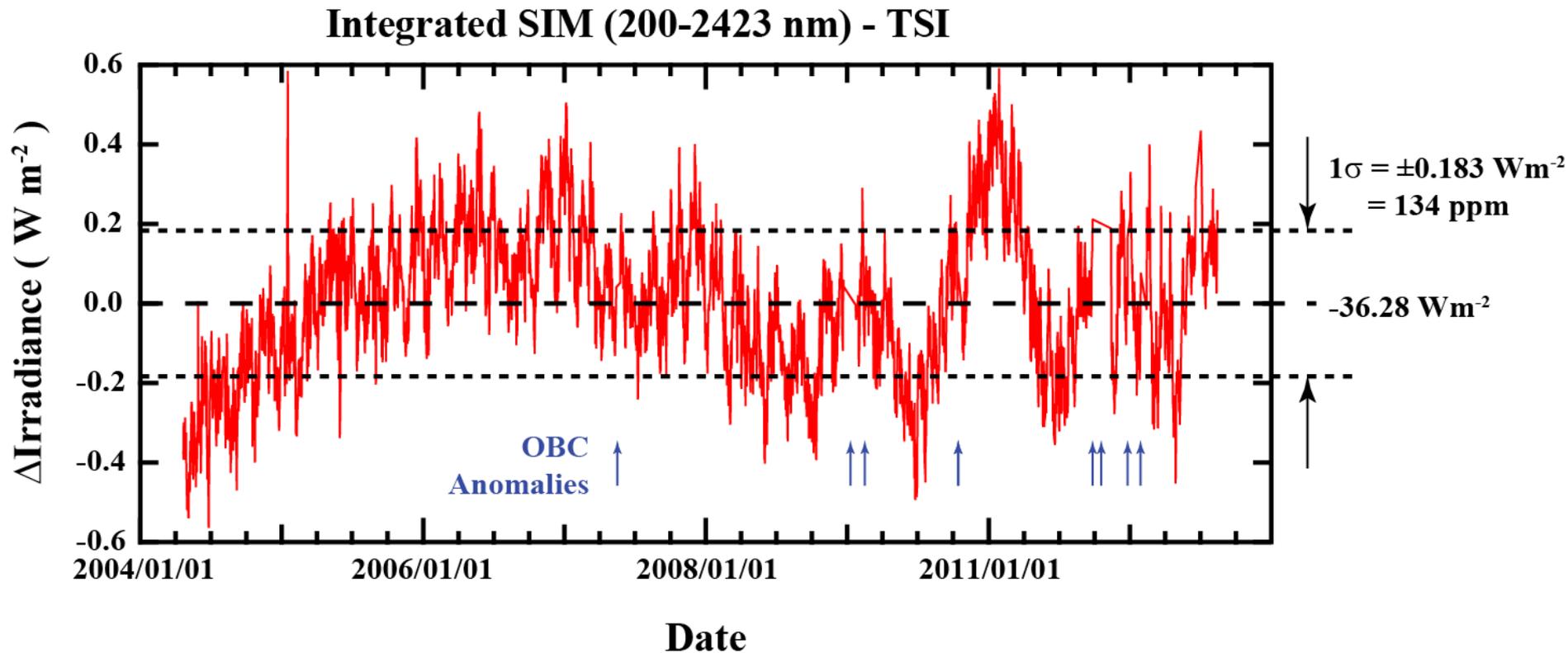


- **Wavelength drive exhibits remarkable precision over very long time periods**
  - Wavelength registration within the limits set by hardware
  - Cardinal wavelength step = 38 subpixel steps, precision  $\sim 1/38$  of a step

# Current SIM Data Composite



- OBC anomalies after 2007 disrupt the continuity of the A-to-B data
- Spectra from the two channels are valid in a piecewise sense
- Connecting data sets after OBC events is a 'compositing' activity and induces discontinuous uncertainty beyond AB comparisons
- Alternate/improved methods are understudy to correct and effectively concatenate the pieces.



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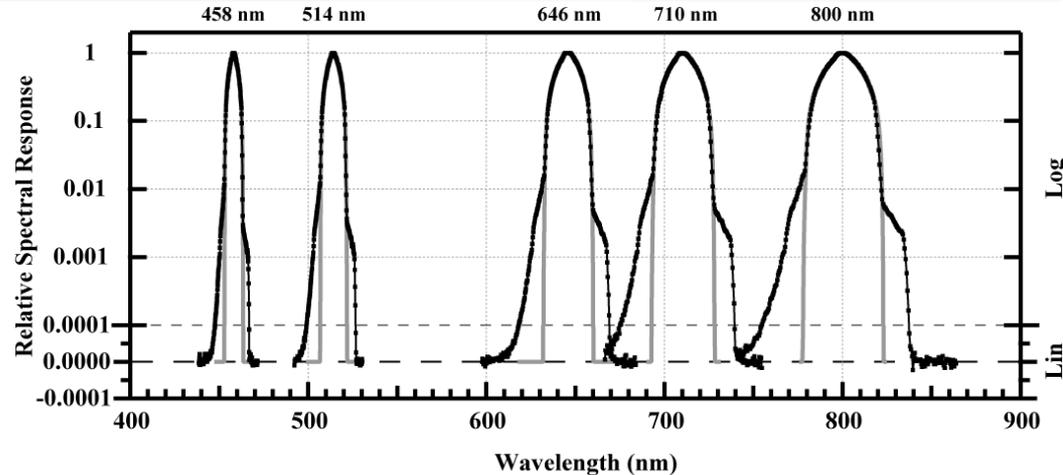
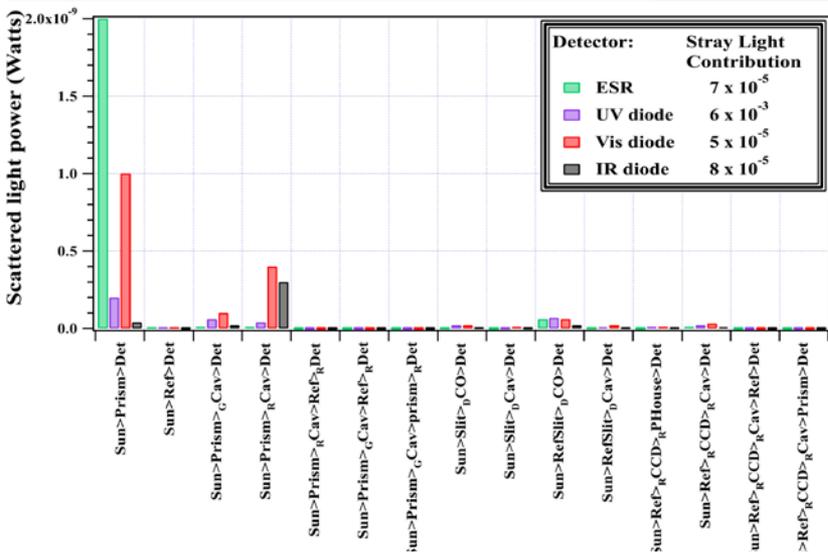
- No systematic slope through the data, suggests no common-mode degradation

- **Long-term degradation corrections in SIM are based solely on measured instrument quantities.**
  - Correction is based on the comparison of two identical (mirror image) spectrometers that have been exposed at different rates.
  - Corrections for photodiode detectors in the same channel are made by comparison with the spectrally flat ESR detector after correcting for the different optical paths through the prism.
  - Safe hold events decrease the precision after 2009, further work is underway to correct the influence of these events.

# Additional Uncertainty Analyses

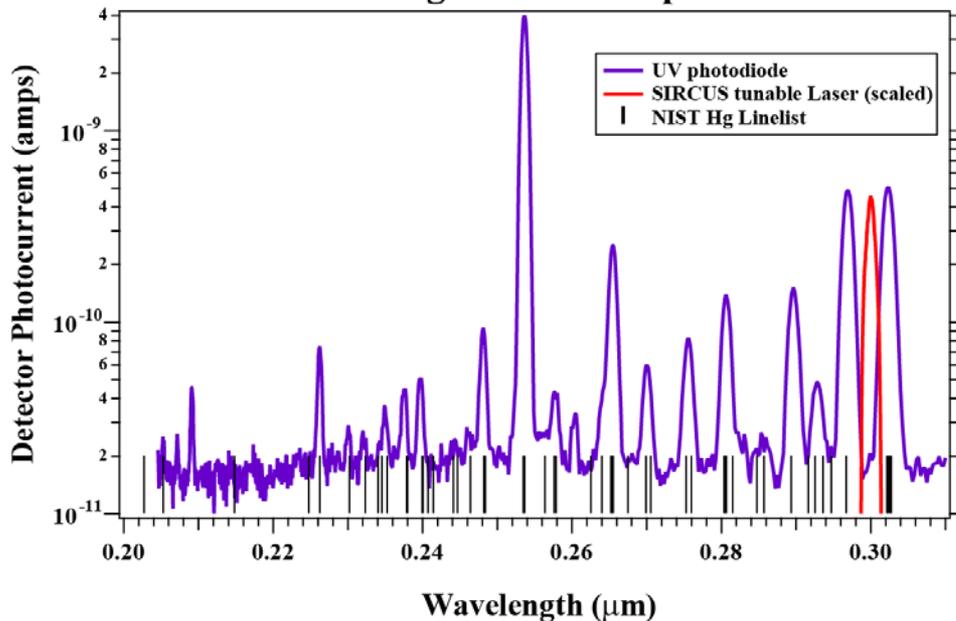
Ranking	Cause	Effect
<i>Short-term effects – do not accumulate with time</i>		
1	Detector Noise	Ultimate limit of comparison of two spectra
2	Spacecraft pointing	Perturbation in prism transmission/wavelength shift
3	Detector Temperature	Spurious structure to photodiode data (700-900 nm range)
4	Prism Temperature	Wavelength shift
5	Scattered light	Increases apparent irradiance – decreases contrast in ‘lines’
<i>Long-term effects – accumulate with time</i>		
1	Prism Transmission Degradation	Long-term reduction in instrument response. Effect is irreversible
2	Photodiode Radiant sensitivity	Reduction in photodiode sensitivity (750-950 nm range)
3	Optical alignment changes	Produces ‘jumps’ in the data at well-defined times
4	ESR servo gain degradation	Reduction in responsivity of the ESR detector

# Short Term effects: Scattered Light



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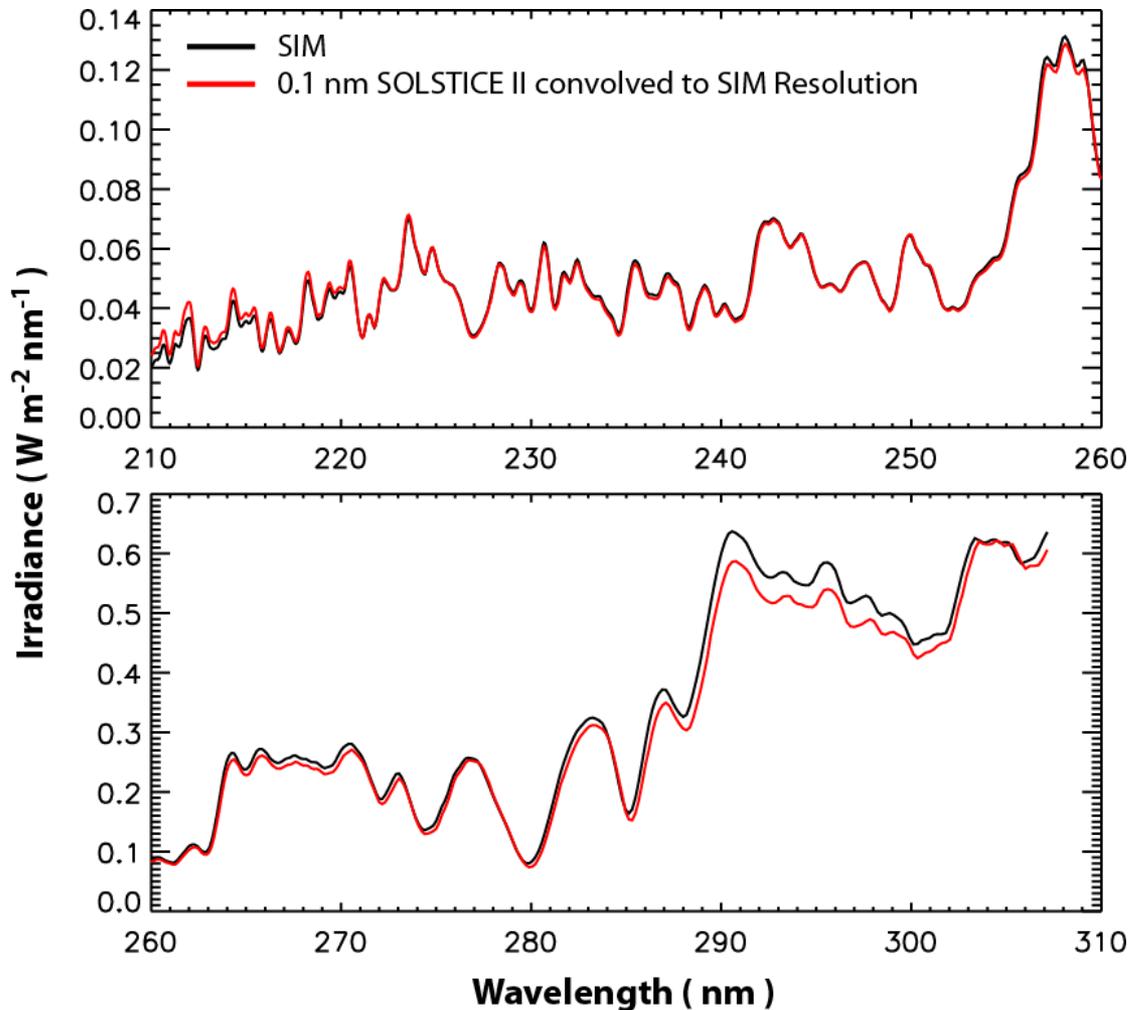
## Hg<sup>198</sup> EDL Lamp



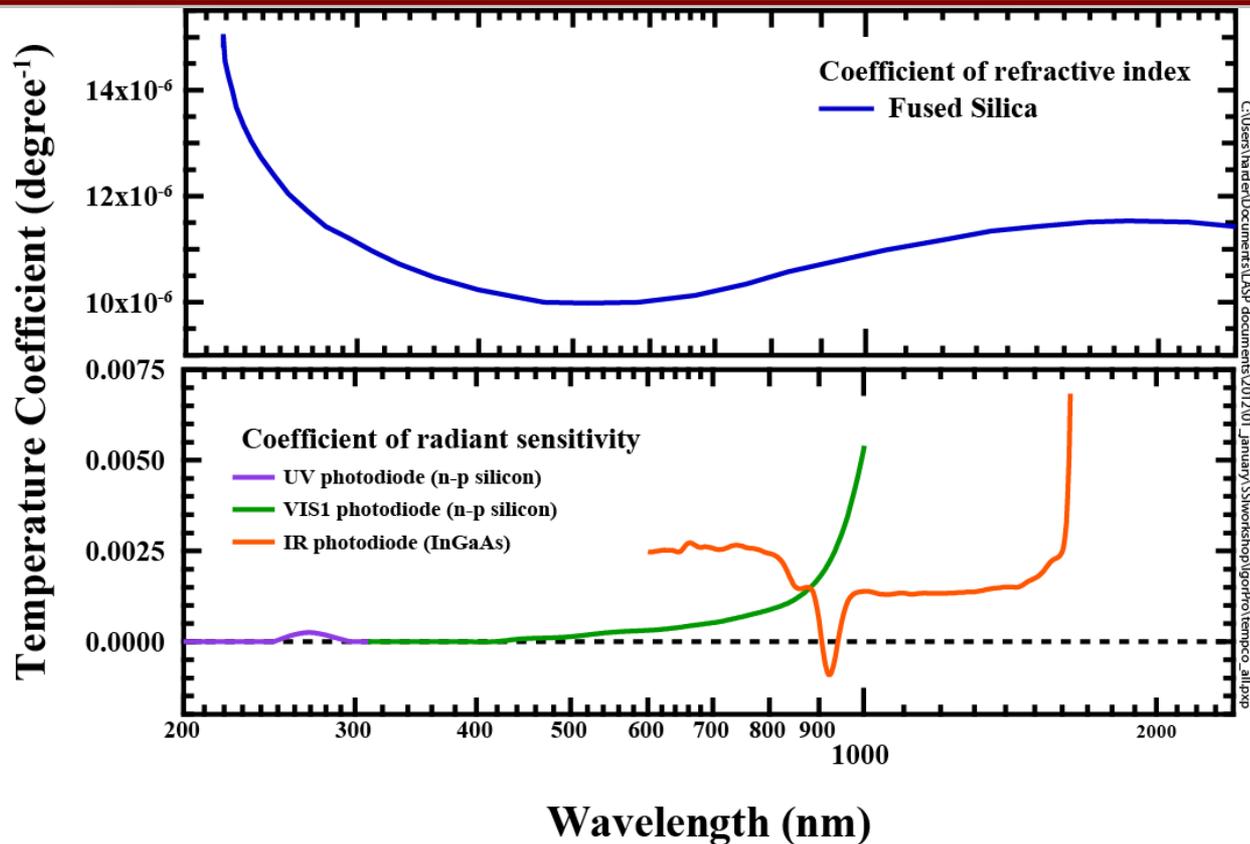
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- Scattered light contribution dominated by polish on prism (near-field scattering)
- Reflections from prism  $\Rightarrow$  cavity  $\Rightarrow$  detector (far-field scattering) are baffled in front of prism & in front of detectors.
- Scattered light cannot be detected above noise level of the instrument. No corrections made in data processing.

# Short Term effects: Scattered Light

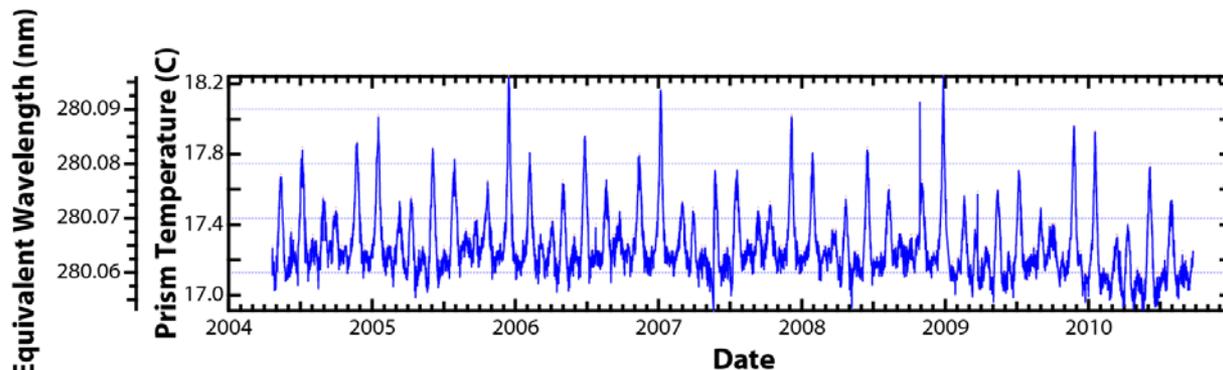


- At the same resolution feature depths match (though absolute scales differ) indicating little discernible scattered light in either instrument.
- Reflects fidelity in determination of the instrument profile integral.

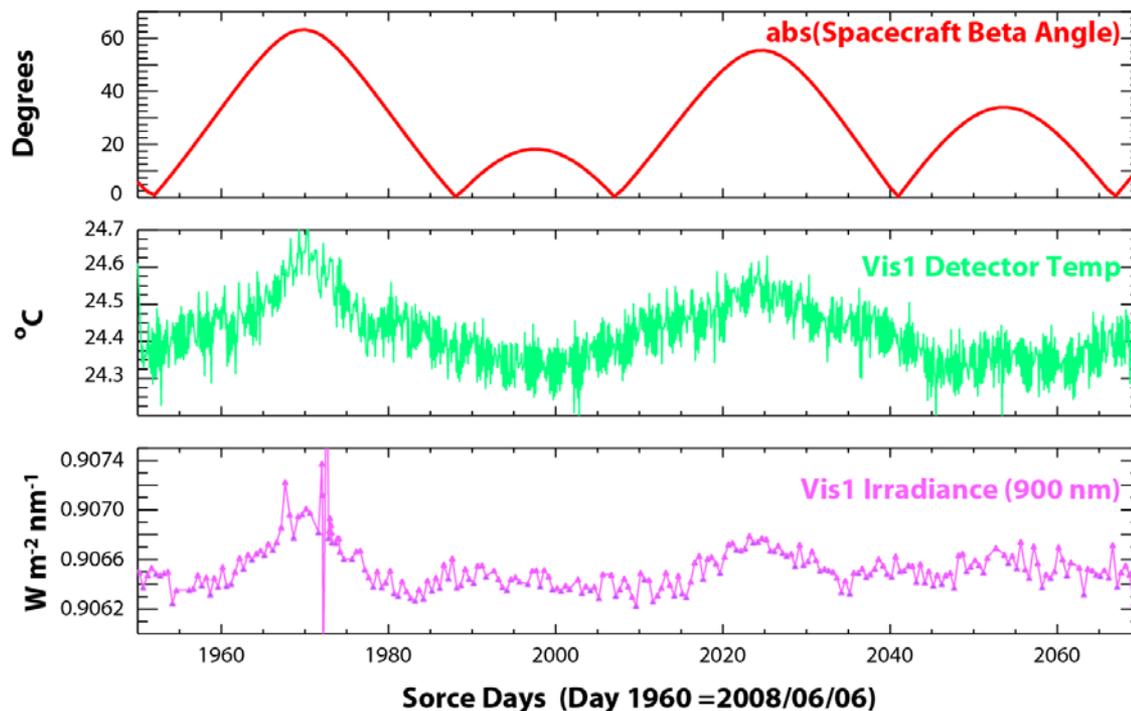


- Index of refraction a function of temperature and effects wavelength reaching the detectors – if not corrected induces wavelength shift
  - Temperature of the prism(s) monitored and corrected continuously in data processing
- Radiant sensitivity is weakly temperature dependent – induces temperature structure in irradiance in the 900-1000 nm range
  - Processing includes a temperature correction, imprecision due to gradients between temperature sensor and the photodiode.

# Short Term effects: Temperature

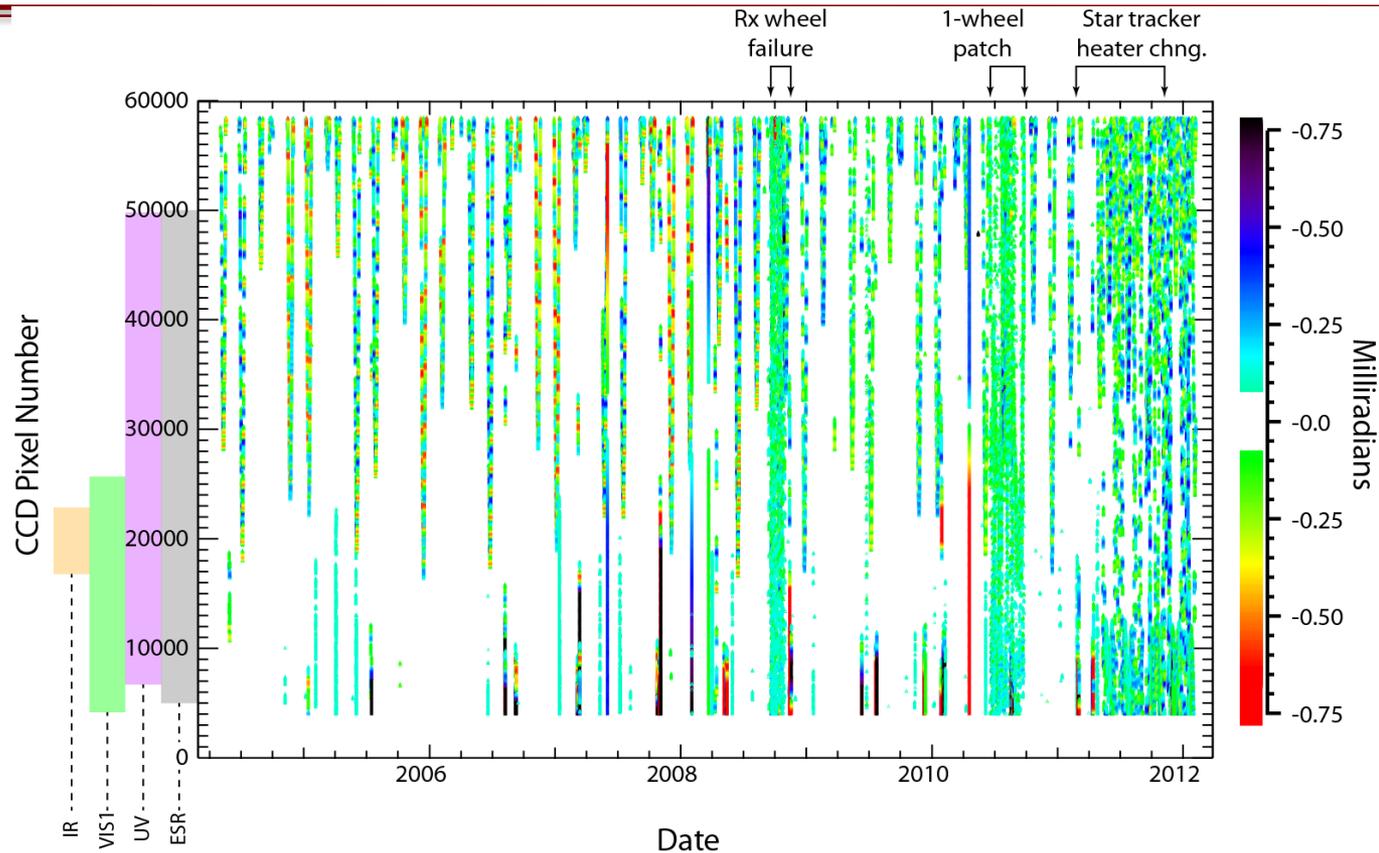


- Prism Temperature variation of  $\sim 1.2$  °C
- P-p  $\lambda$  shift  $\sim 0.04$  nm
- less than 1/4 of cardinal prism step.
- Corrected in data processing



- Residual structure in irradiance related to inaccurate corrections for diode temperature
- Adds uncertainty of  $< 0.1\%$  to determining long-term trend in the 900-1000 bands.

# Short Term effects: Spacecraft Pointing



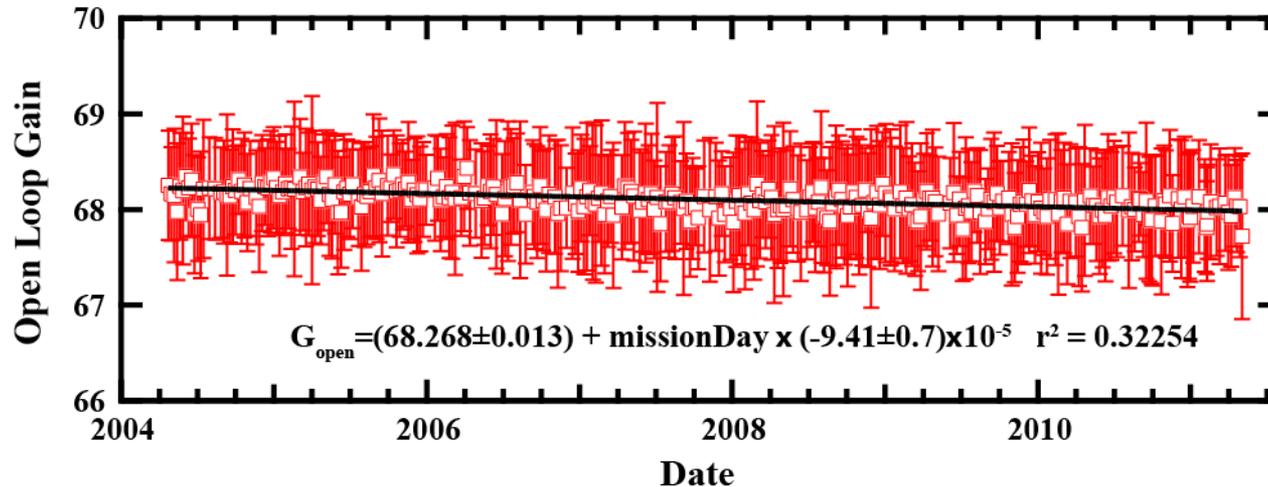
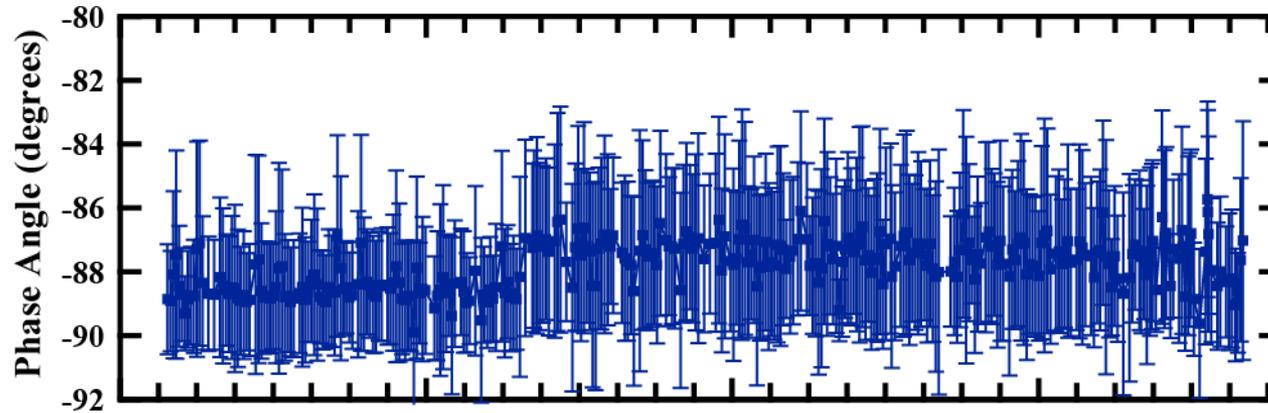
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- Roll maneuvers contaminate only 2.8% of the viable science data
- Effects of rolls tend to preferentially effect data towards the end of scans but can effect any portion of the scan.
- Not all 'spikes' in the data are identifiable as caused by roll maneuvers and vice versa
  - 'De-spiking' more practical than filtering: no correction made in data processing

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<i>Long-term effects – accumulate with time</i>		
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2	<b>Optical alignment changes</b>	<b>Produces ‘jumps’ in the data at well-defined times</b>
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4	<b>ESR servo gain degradation</b>	<b>Reduction in responsivity of the ESR detector</b>

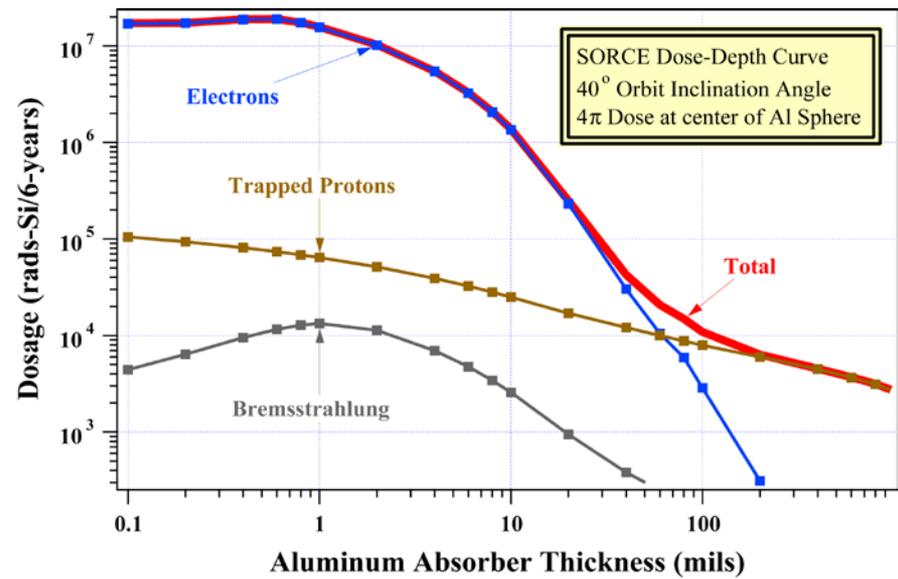
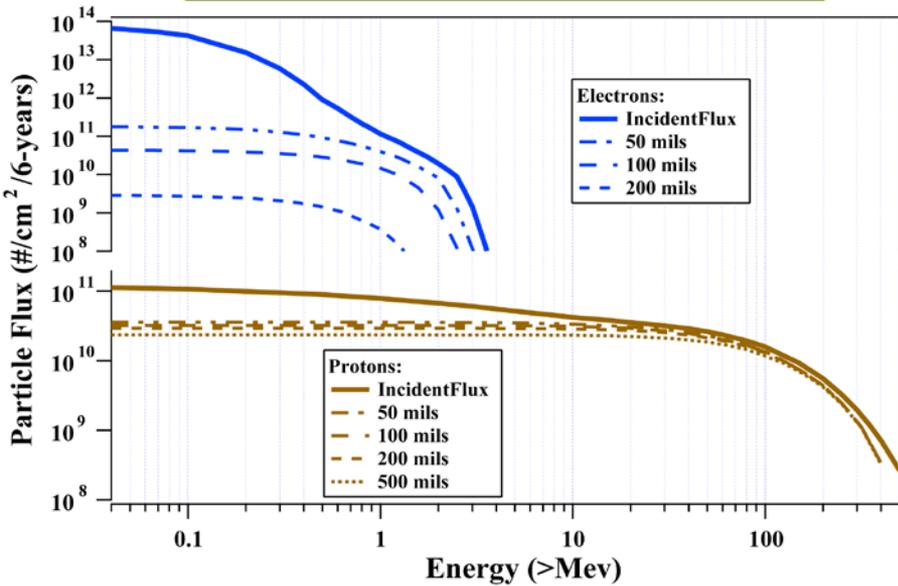
# Long Term effects: ESR Gain



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- Gain change discernable, but small
- Change in SIM B comparable to SIM A
  - $G_{\text{open B}} = (68.541 \pm 0.018) + (-7.78 \pm 0.86) \times 10^{-5} \quad r^2 = 0.16238$
- Corrections made in data processing

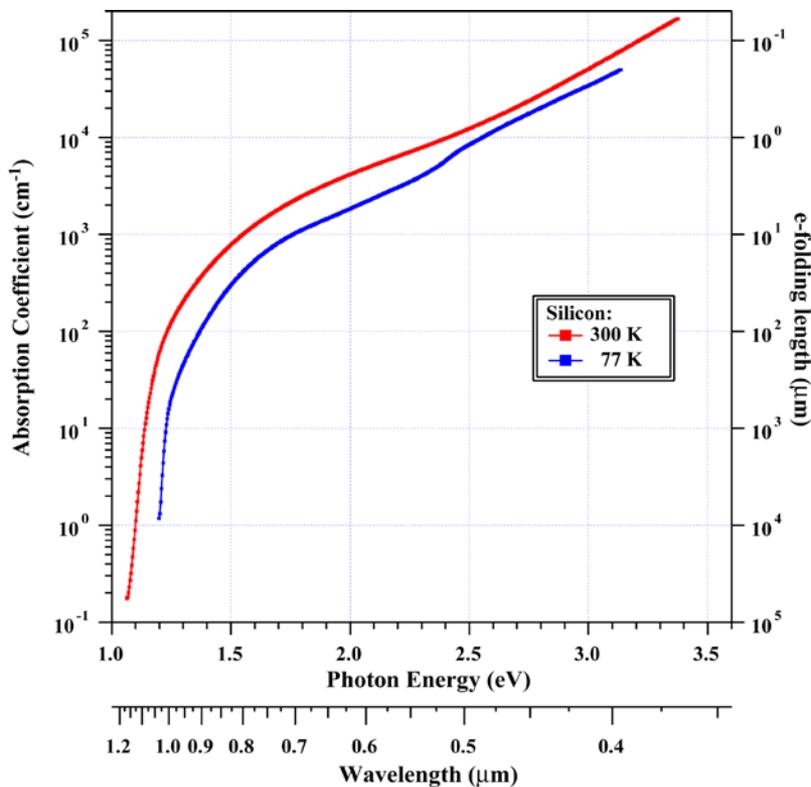
Orbit Inclination = 40° Height = 650/650 km 6 years (4 Max + 2 Min)



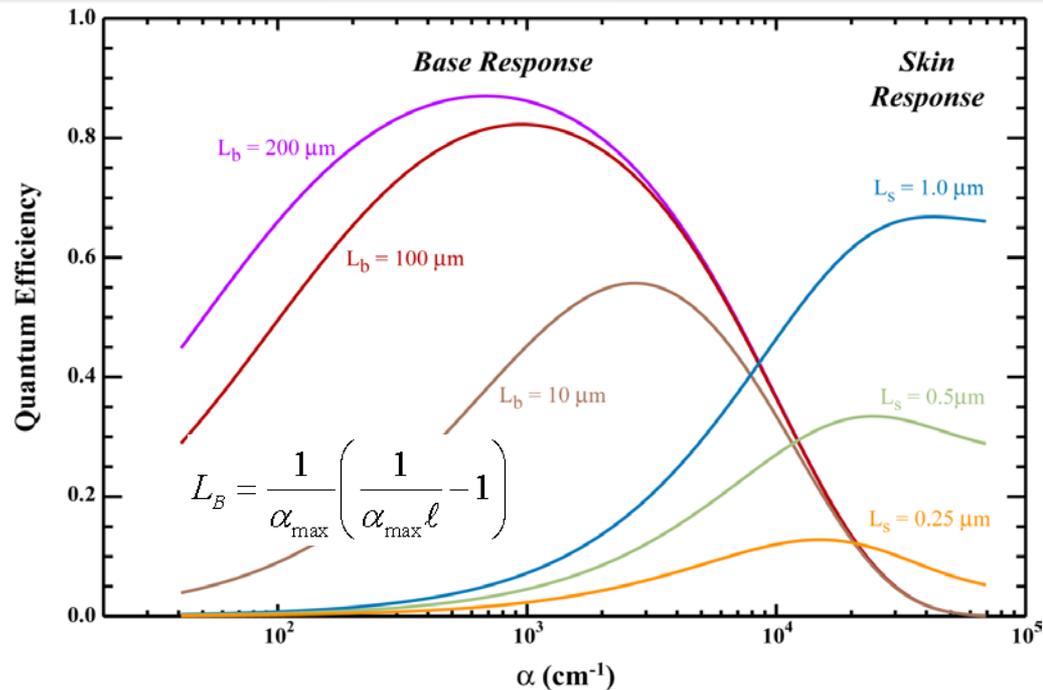
- **High Energy Particle Spectrum**
  - High energy particles (particularly protons) penetrate deep into the instrument case

- **The majority of high energy particles will be protons**
  - Shielding ineffective against proton penetration

# Long Term effects: Photodiode Radiant Sensitivity changes



adapted from: Dash and Newman, *Phys. Rev.*, **99**, 1151, 1955



J. A. Baicker and B. W. Faughnan, *J. Appl. Phys.*, **33**, 1962.

- Radiation damage in the *base* of the diode decreases  $L_b$  (minority carrier diffusion length) and effectively moves  $\alpha_{\max}$  to shorter wavelengths (larger  $\alpha$ )
- $L_{\text{skin}}$  dominates the diodes performance at the shorter wavelengths.
- Radiation testing suggests p-n diode incurs greater damage than n-p:
  - $\sim 100x >$  for electrons,  $\sim 3x >$  for protons

# SIM Measurement Equation and Error Budget

$$E(\lambda_s) = \frac{P_{ESR}(\lambda_s)}{A_{slit} \int \alpha_\lambda T_\lambda \Phi_\lambda S(\lambda, \lambda_s) d\lambda} \quad \text{or} \quad E(\lambda_s) = \frac{P_{detector}(\lambda_s)}{A_{slit} \int R_\lambda T_\lambda \Phi_\lambda S(\lambda, \lambda_s) d\lambda}.$$

Term (units)	Symbol	Value/Range	Uncertainty	Derived from:
Wavelength (nm)	$\lambda$	265-2423	$0.2 \pm \lambda \times (150 \times 10^{-6})$	$\lambda$ standards, solar spectrum
Power (watts) of the ESR	$P$	$1 \times 10^{-7} - 5 \times 10^{-5}$	$\sim 2 \times 10^{-9} \text{ WHz}^{-1/2}$	Detector testing
Entrance slit area (mm <sup>2</sup> )	$A_{slit}$	2.1	$5 \times 10^{-5}$	Slit diffraction
ESR optical efficiency (%)	$\alpha_\lambda$	100	+0 to -2 (200-1000 nm) +0 to -10 (1000-2700 nm)	SIRCUS, flight spare ESR (measured in power mode)
Photodiode radiant sensitivity (amps/watt)	$R_\lambda$	0.08 - 1.0	2-4% (wavelength dependent)	Comparisons with ESR
Prism transmission (%)	$T_\lambda$	0.55-0.77	$\pm 0.1\%$ 200-700 nm $\geq \pm 1\%$ 700-2700 nm	Laboratory measurements (see Harder et al, 2005a)
Diffraction loss (%)	$\Phi_\lambda$	0.3-2.2	$\sim 0.01$	Diffraction theory
Instrument function area (nm)	$S$	0.58-34.5	$\sim 0.4\%$	Ray tracing, laser scans

See Harder *et al.*, *Solar Physics* (2005): **230** 169-204

# Time series of solar spectral variability from **SORCE**

