

Mars Atmosphere and Volatile
Evolution Mission
Science Community Workshop

Science Overview

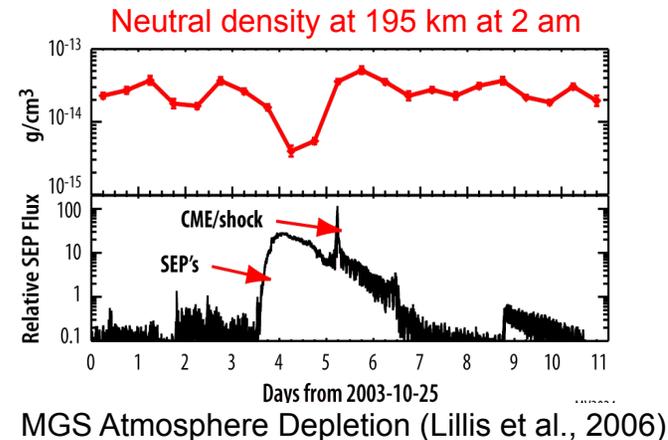
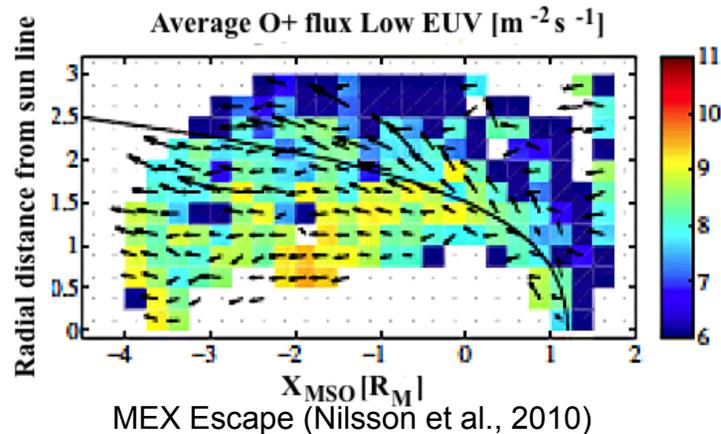
J. M. Grebowsky

MAVEN Community Workshop, Dec 3, 2012 San Francisco, CA

Determine The Rate Of Escape Today and the History of Loss Through Time.

Evidence for the Loss to Space Today

- Detection of energetic ionospheric ions moving away from the planet by *Mars Express* and *Phobos* Missions
- Mars Global Surveyor inferred observations of decrease in upper atmosphere density in response to a Solar Energetic Particle event

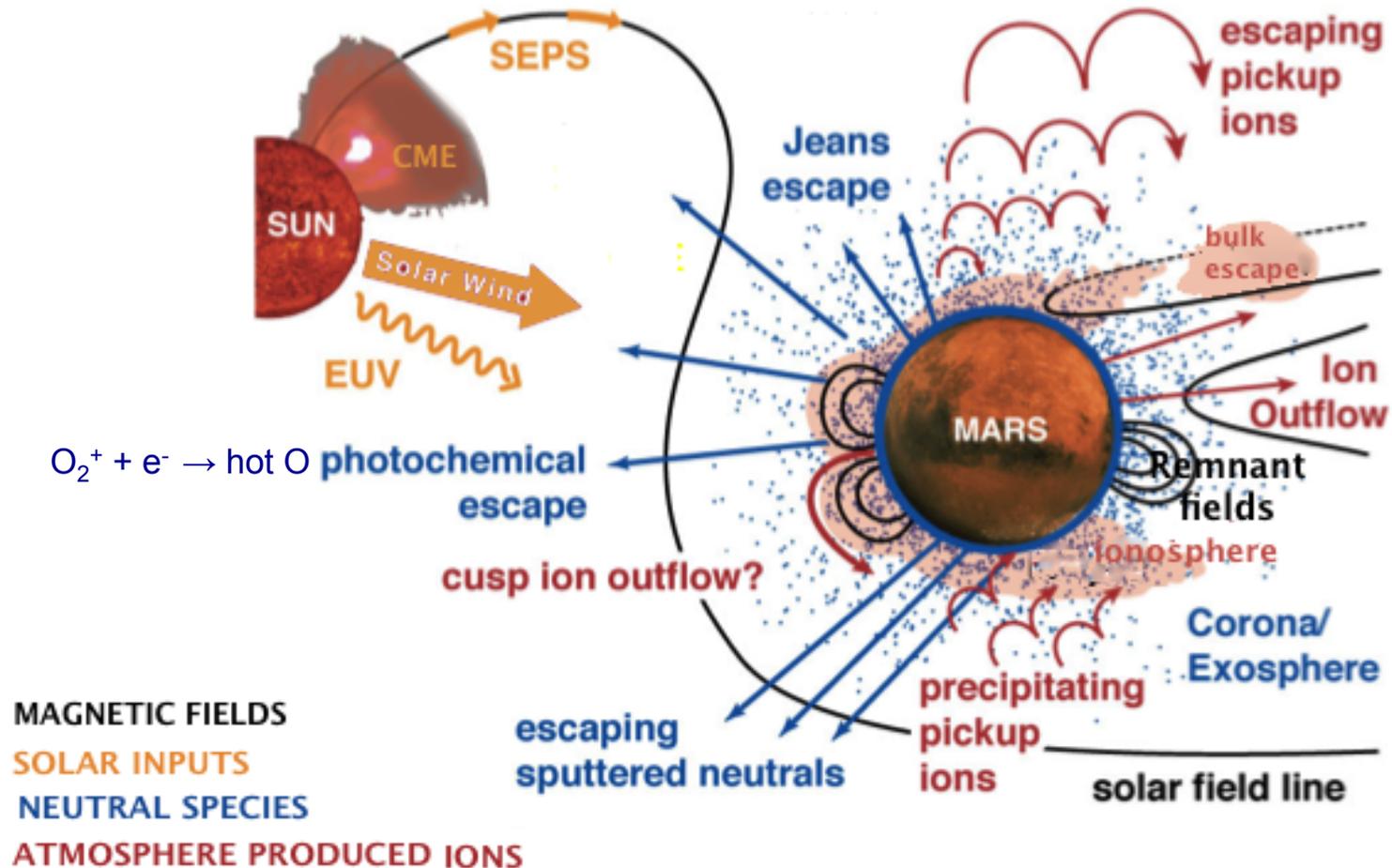


Variation of Escape Through Time

- Absence of water
- Current atmospheric Isotopic ratios favor lighter species compared to cosmic abundances.
- Ion escape fluxes dependent upon solar evolution - at solar max (Phobos) ~ 10X larger ion escape than during solar minimum (Mars Express)

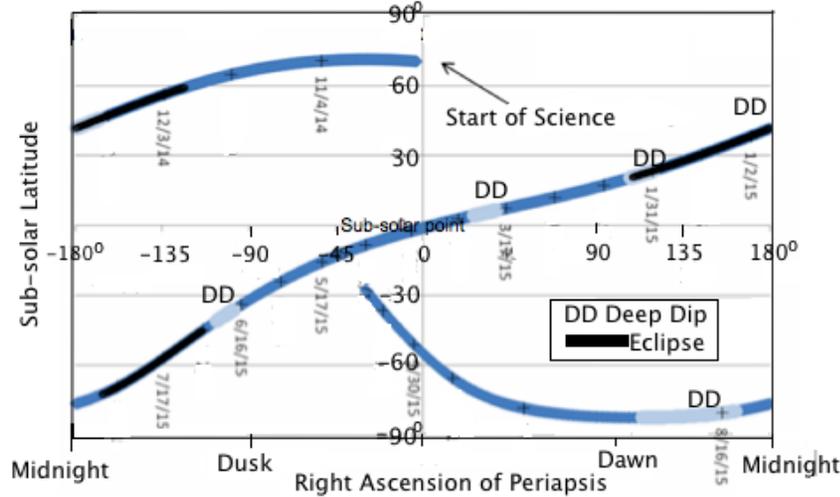
- Thermal (Jean' s) escape : e.g. H
- Non-thermal escape:
 - (1) *Photochemistry* : Dissociative recombination of O_2^+ , N_2^+ , CO^+ with e^- forming energetic (hot) atoms (e.g. O, N, C) with escape energies
 - (2) *Pick-up ionization* : ions produced in the extended atmosphere (photoionized or charge exchange) are dragged along by solar wind electric field to partially escape (O^+ , H^+ , C^+ ...)
 - (3) *Ionospheric outflows*: planetary ions are accelerated by SW induced ionospheric EM-fields and lost in the wake (e.g. O_2^+)
 - (4) *Ion sputtering*: a portion of SW and pick-up ions impact the neutral atmosphere with enough energy to eject neutral atmospheric particles at/above the exobase (e.g. CO_2 , N_2 , CO , O, N, C...).

Magnetic Field Once Protected Atmosphere Now, Solar Wind/IMF Penetrates the System

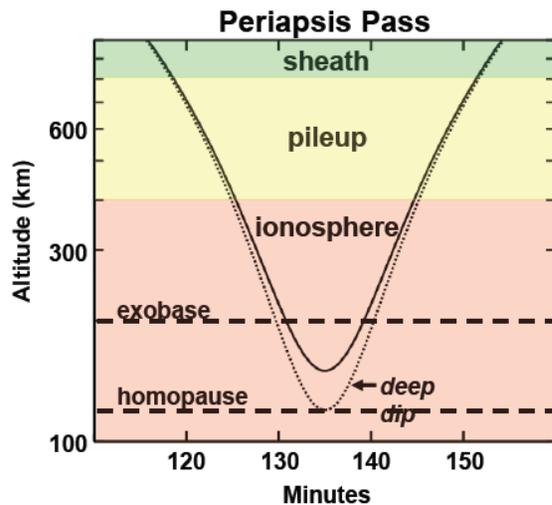
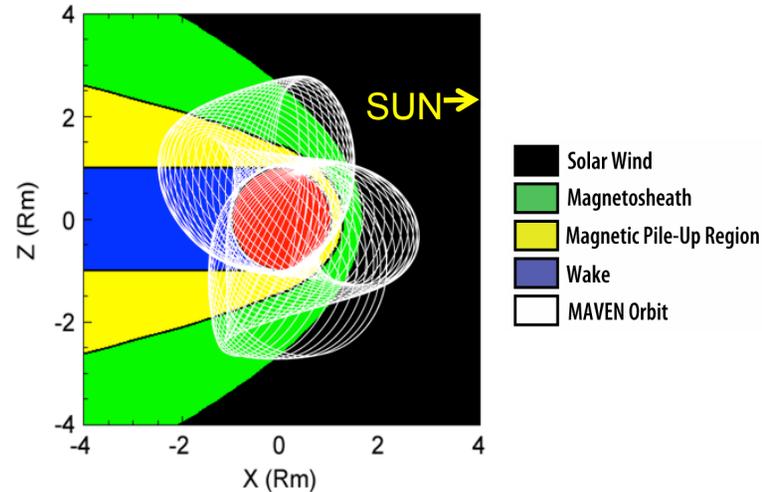


Orbit was Judiciously Selected

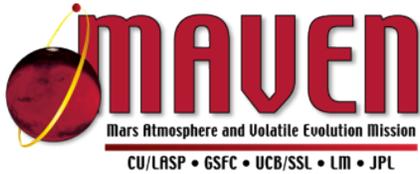
Periapsis Coverage of Latitude-Local Time



Looking Down on Polar Cap



- Continuous orbit-to-orbit measurements
- Comprehensive diurnal coverage and periapsis passages through SH remnant magnetic field zones
- Traversal of all important magnetosphere – solar wind regions (apoapsis in solar wind; periapsis below exobase)
- Five Deep-dip campaigns to explore noon, midnight, polar cap and terminator regions.



MAVEN Measurement Aims

Reservoirs

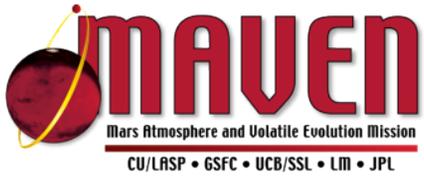
- Characterization of upper atmospheric composition, structure, variability
- In situ measurements down to below the exobase
- Measurement of ionospheric thermal state

Escape

- Measurement of ions from low (non-escaping) to high energies
- Track escaping ions down to exobase, to constrain driving loss processes
- Determination of relationship of state properties and of escape to locally magnetized regions and to direction of IMF
- Measurements of all charged particle populations with improved measurements compared to the pathfinder MEX mission (e.g. improved angular resolution, energy range, and pitch angle measurements for all instruments)

Evolution

- Measurement of isotope ratios that relate to atmospheric evolution
- Simultaneous measurements of energy inputs (EUV, E/M waves, Solar Wind, Solar Energetic Particles) to correlate escape variation with solar changes



The Measurements



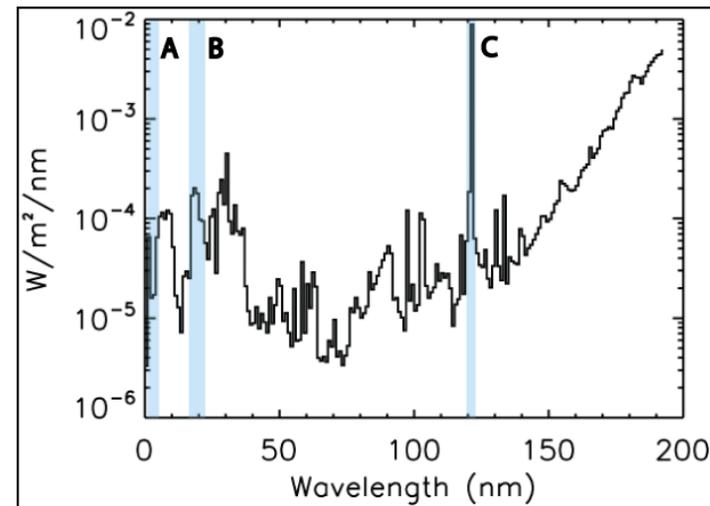
EUV MONITOR
FLIGHT MODEL

Measurement Objectives:

- Solar EUV irradiance variability at wavelengths important for ionization, dissociation, and heating of the upper atmosphere (wavelengths shortward of H I Ly- α 121.6 nm)

Technical details and heritage:

- Three photometers at key wavelengths representing different temperature solar emissions (0.1-7, 17-22, and 121.6 nm)
- Heritage from TIMED, SORCE, SDO and rocket instruments
- Full spectrum (0-200 nm) derived from measurements using Flare Irradiance Spectral Model (FISM).



Langmuir Probe and Waves (LPW)

Bob Ergun, LASP



Stacer booms to be used for deployment

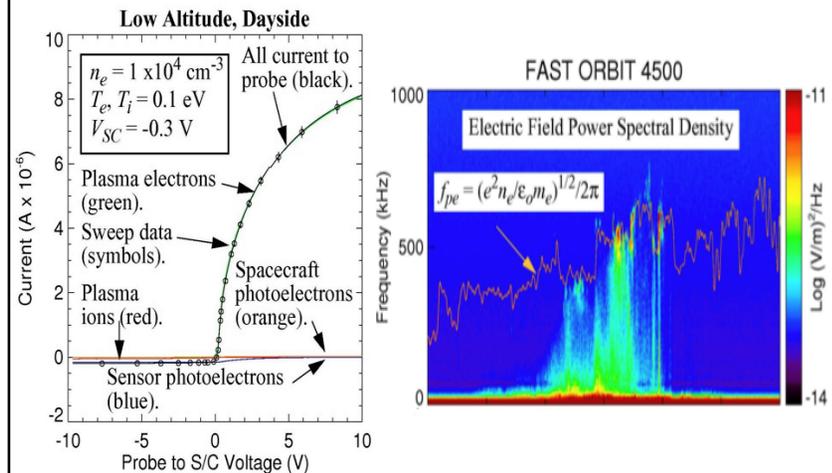
Measurement Objectives:

- Electron temperature and number density throughout upper atmosphere
- Electric field wave power at low frequencies important for ion heating
- Wave spectra of naturally emitted and actively stimulated Langmuir waves to calibrate density measurements

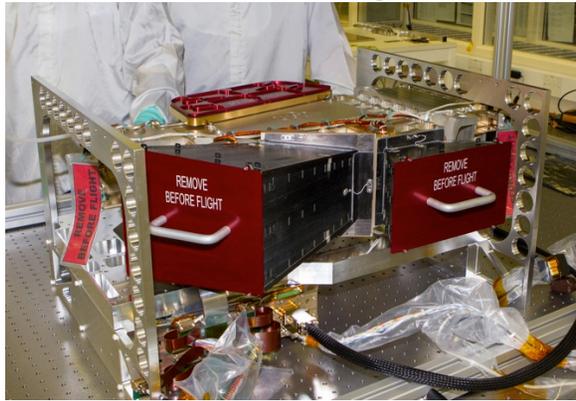
Includes EUV monitor, described earlier

Technical details and heritage:

- Cylindrical sensors on two 7-meter booms
- Sensor I-V sweeps (at least ± 50 V range)
- Low frequency (f : 0.05-10 Hz) E-field power
- E-Spectra measurements up to 2 MHz
- White noise (50 kHz-2 MHz) sounding
- Heritage from THEMIS and RBSP



IUVS and DPU Flight Models

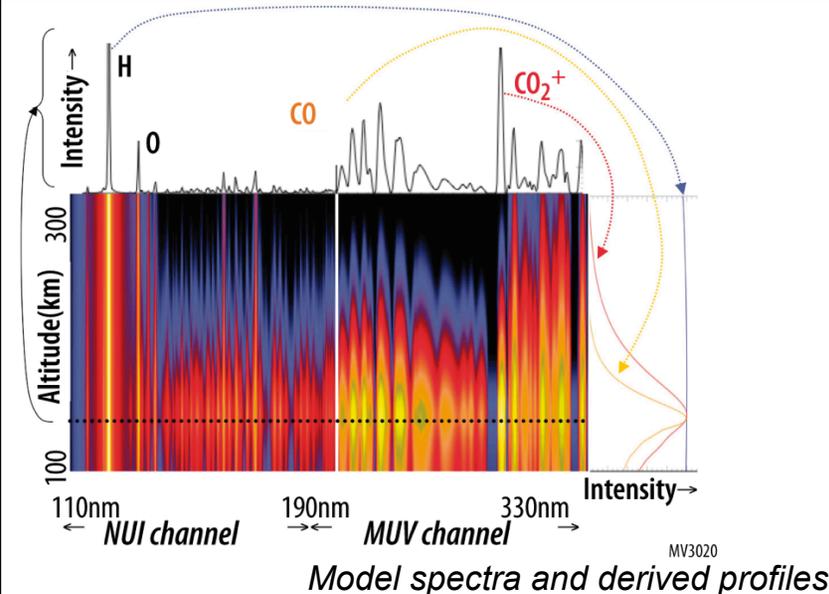


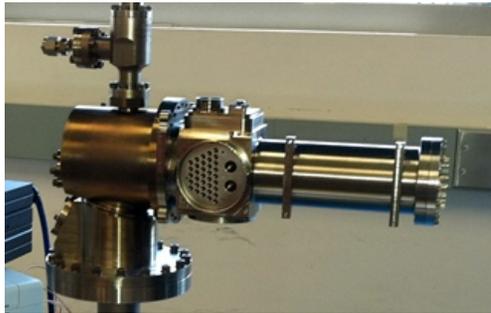
Measurement objectives:

- Vertical profiles of neutrals and ions through limb emissions and stellar occultations.
- Disk maps from near apoapsis.
- D/H and hot oxygen coronal mapping.
- Atmospheric properties below homopause.

Technical details and heritage:

- Imaging spectroscopy from 110-340 nm, with resolution of 0.5 – 1.0 nm.
- Vertical resolution of 6 km on limb, horizontal resolution of 200km
- Most recent heritage from AIM CIPS





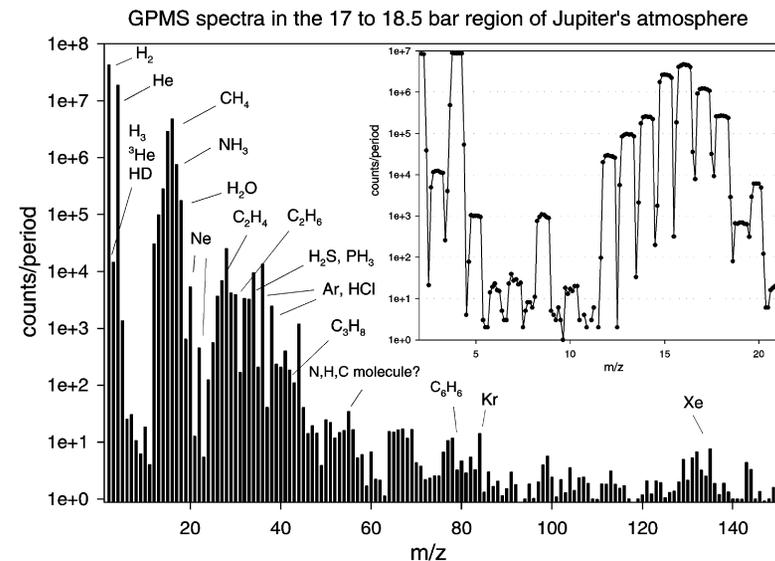
NGIMS Quadrupole Mass Spectrometer Sensor

Measurement Objectives:

- Basic structure of the upper atmosphere (major species He, N, O, CO, N₂, NO, O₂, Ar and CO₂) and ionosphere from the homopause to above the exobase
- Stable isotope ratios, and variations

Technical Details:

- Quadrupole Mass Spectrometer with open and closed sources.
- Closed source: non-reactive neutrals
- Open source: neutrals and ions
- Mass Range: 2 - 150 Da
- Heritage: Galileo GPMS, Pioneer Venus ONMS, CASSINI INMS, Contour NGIMS



Accelerometer Science (ACC)

Rich Zurek (JPL)



Inertial Measurement Units (IMU)
[Accelerometers]

Measurement Objectives:

Vertical profiles of atmospheric density and temperature; possible wind profiles

Technical Details:

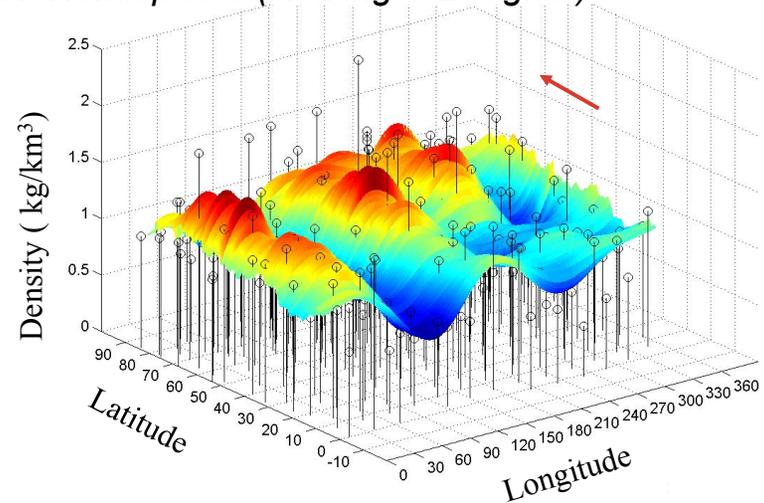
- Density derived from acceleration data a_d

$$\rho = \frac{2ma_d}{C_d S v^2}$$

$m, S, v, \text{ \& } C_d = \text{spacecraft mass, X-section, velocity, \& drag coefficient}$

- Winds derived from NAV and ACS data

Atmospheric Structure: *Odyssey saw large amplitude waves during early morning in the N. Hemisphere (Keating & Bougher)*



Magnetometer Sensors



Measurement objectives:

- Vector magnetic field in the unperturbed solar wind ($B \sim 3$ nT), magnetosheath ($B \sim 10$ -50 nT), and crustal magnetospheres ($B < 3000$ nT), with the ability to spatially resolve crustal magnetic cusps (horizontal length scales of ~ 100 km).

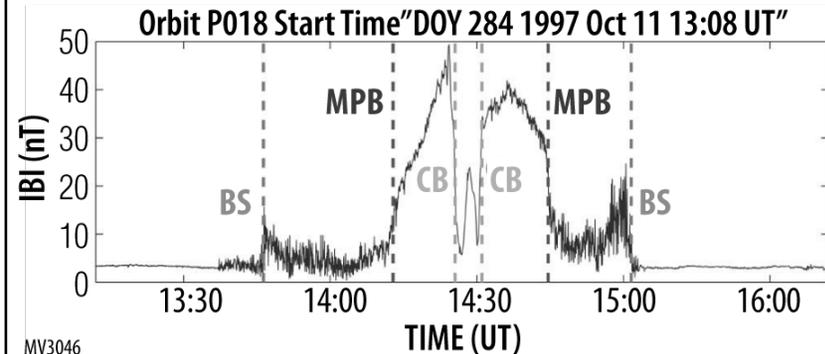
Technical details and heritage:

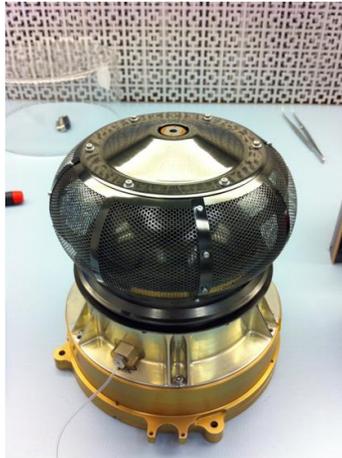
-2 sensors, outboard of solar array

-Magnetic field over a dynamic range of $\sim 60,000$ nT; resolution 0.05 nT

-Heritage: *MGS, Voyager, AMPTE, GIOTTO, CLUSTER, Lunar Prospector, MESSENGER, STEREO, Juno, RBSP*

MGS MAG measurements:





SWEA Flight Model

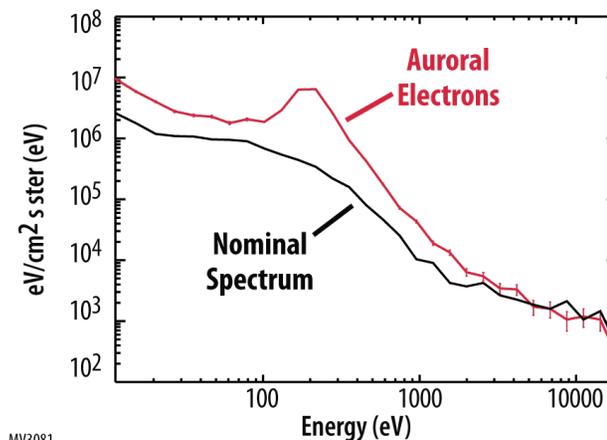
Measurement objectives:

- Measure energy and angle distributions of electrons in the Mars environment
- Determine electron impact ionization rates
- Measure magnetic topology via loss cone measurements
- Measure primary ionospheric photoelectron spectrum
- Measure auroral electron populations
- Evaluate plasma environment

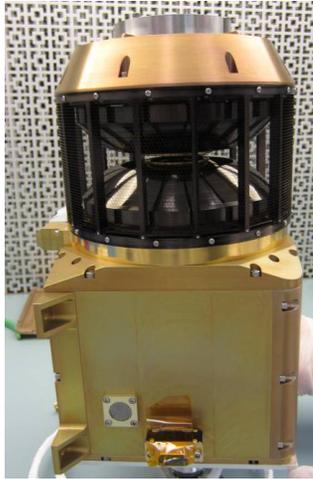
Technical details and heritage:

- Hemispherical Electrostatic Analyzer
- Electrons with energies from 5 eV to 5 keV
- FOV 360° X 130°
- Based on STEREO SWEA.

MGS measurements of auroral electrons:



MV3081



Flight Model

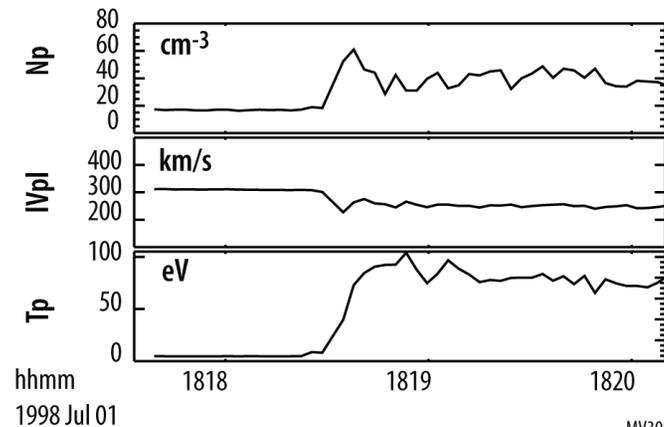
Measurement objectives:

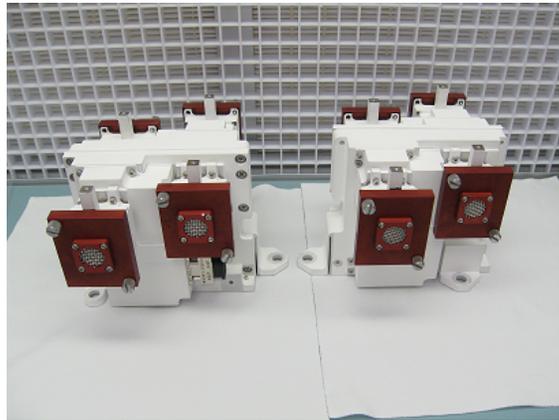
- Density and velocity distributions of solar wind and magnetosheath ions to determine the charge exchange rate and the bulk plasma flow from solar wind speeds (~350 to ~1000 km/s) down to stagnating magnetosheath speeds (tens of km/s).

Technical details and heritage:

- Proton and alpha particle velocity distributions from <50 to >2000 km/s, density from 0.1 to >100 cm⁻³
- Heritage from Wind, FAST, and THEMIS.

Similar measurements provided by *Wind*:





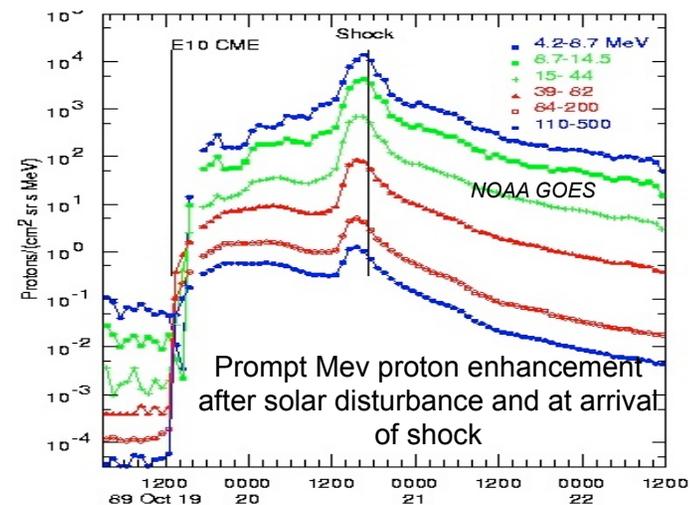
Flight Models

Measurement objectives:

- Characterize solar particles in an energy range that affects upper atmosphere and ionospheric processes (~120 – 200 km)

Technical details and heritage:

- Two dual double-ended telescopes
- Protons and heavier ions from ~25 keV to 12 MeV
- Electrons from ~25 keV to 1 MeV
- Nearly identical to SST on THEMIS.





Flight Model

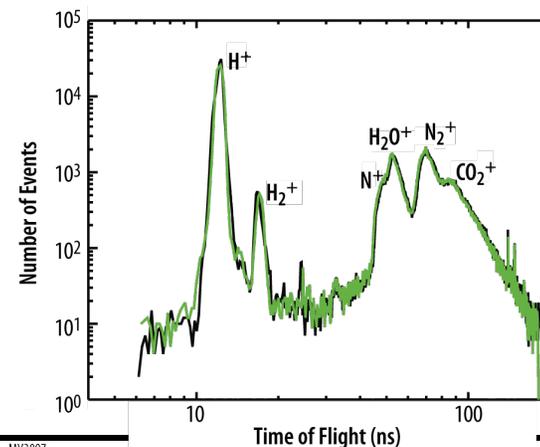
Measurement objectives:

- Escaping ions and processes
- Composition of thermal to energetic ions; energy distributions and pitch angle variations
- Ionospheric Ions 0.1-10 eV
- Tail Suprathermal ions (5-100eV)
- Pick-up Ions (100-20,000 eV)
- Key ions H^+ , O^+ , O_2^+ , CO_2^+

Technical details and heritage:

- Toroidal Electrostatic Analyzer with Time of Flight section
- Mass Range 1-70 AMU, $\Delta M/M > 4$
- Energy range ~ 1 eV to 30 keV, $\Delta E/E \sim 15\%$
- FOV $360^\circ \times 90^\circ$
- Heritage from Cluster CODIF.

Laboratory spectrum from Engineering Development Unit

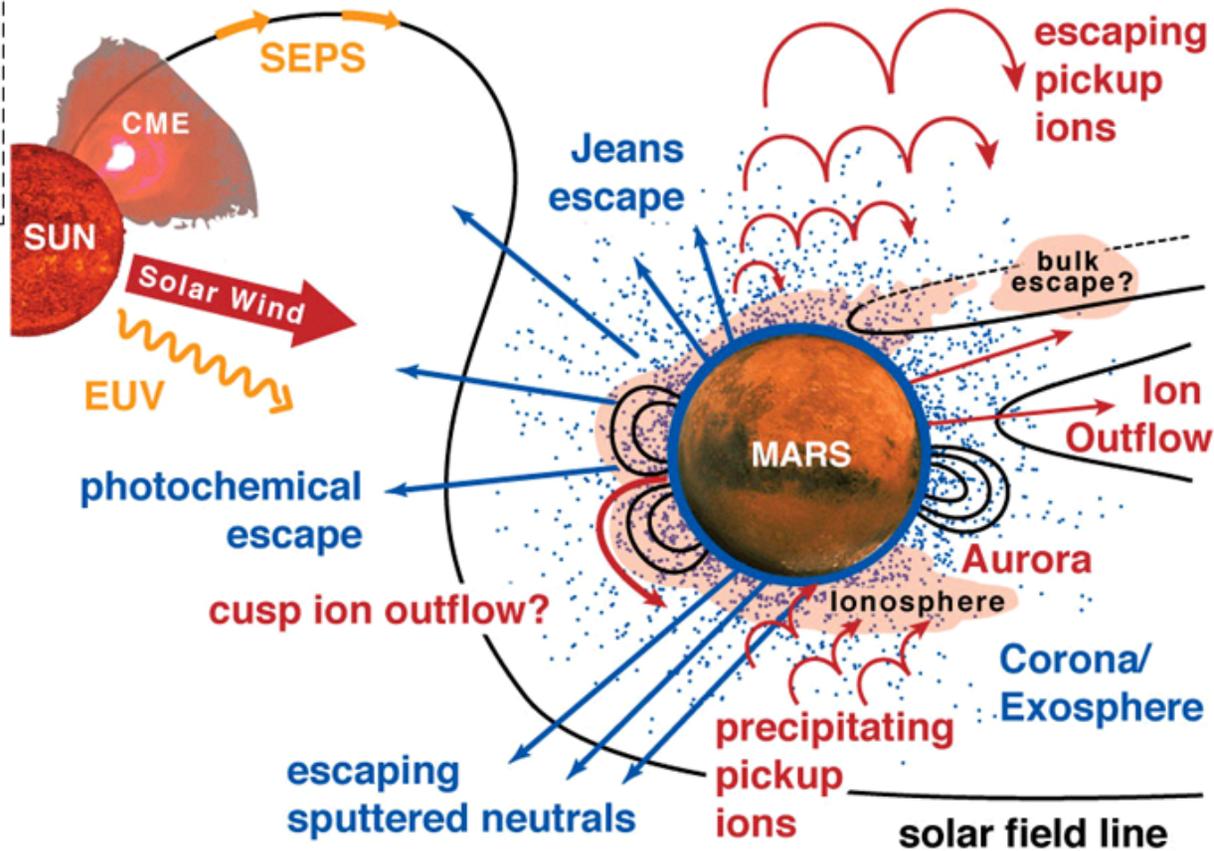


Instruments Sample all the Relevant Physics

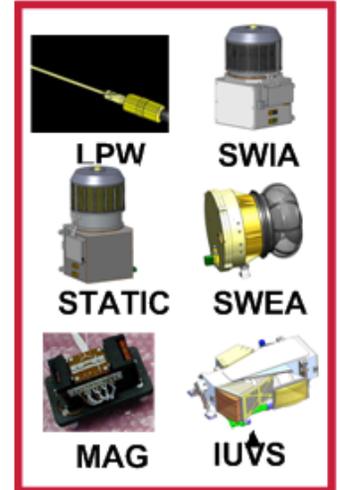
Solar Inputs



- LPW
- EUV Sensor
- SEP
- SWIA
- SWEA
- MAG



Plasma Processes



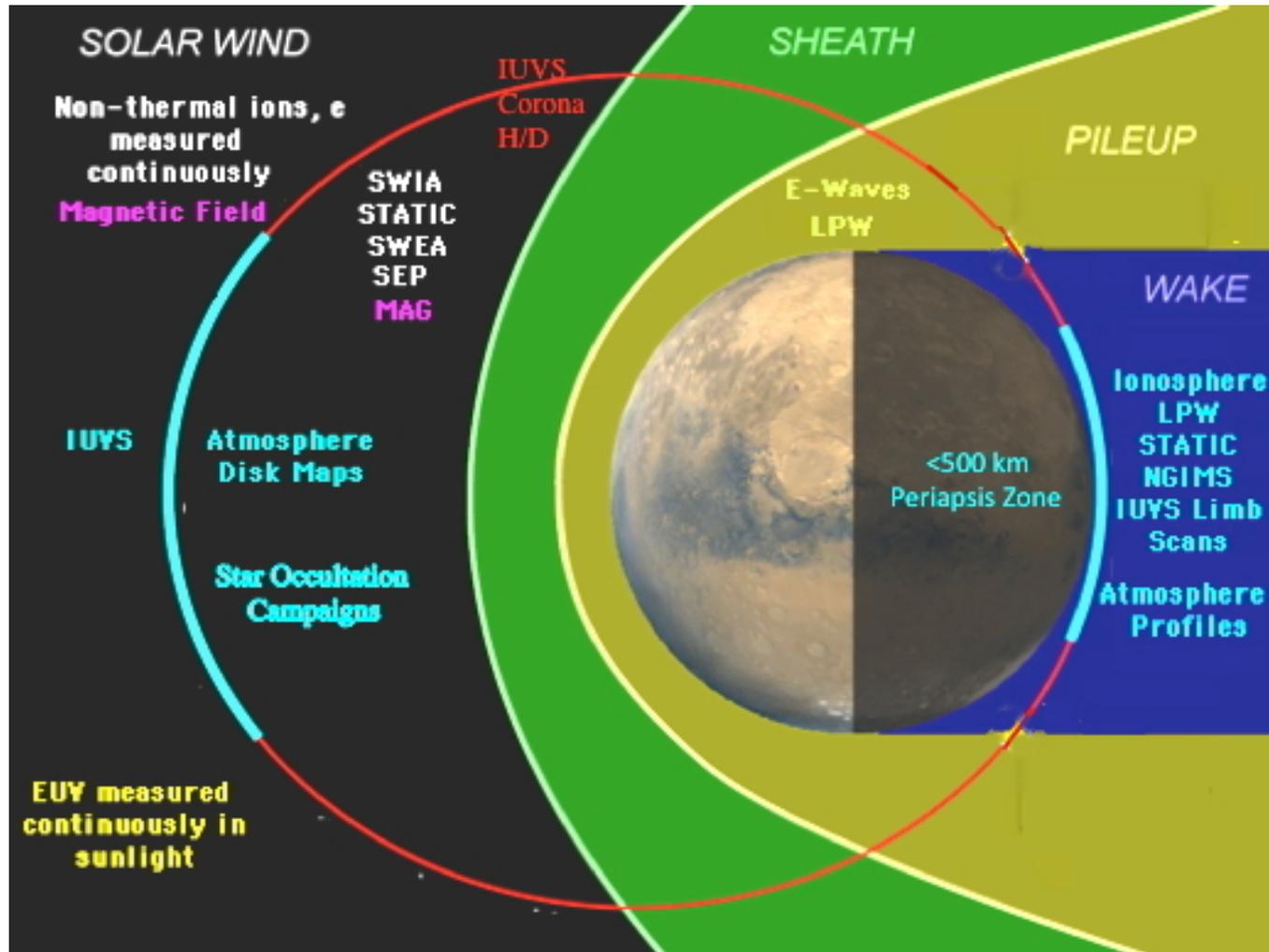
- LPW
- SWIA
- STATIC
- SWEA
- MAG
- IUVS

Neutral Processes

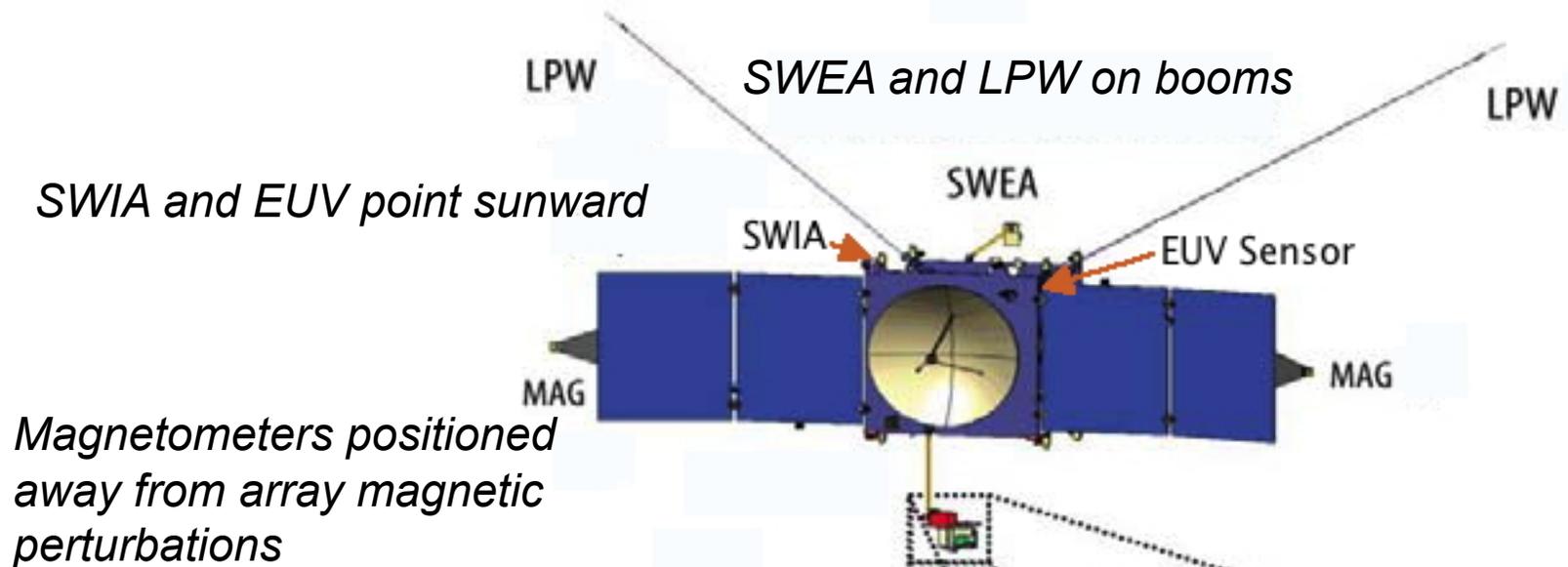


- NGIMS
- IUVS

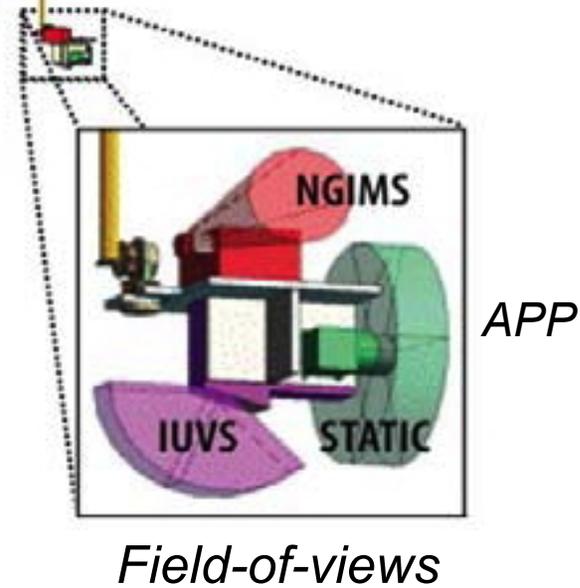
Integrated Sequencing of Measurements



Instrument Placements



- Three instruments on Articulated Payload Platform (APP) allow:
 1. IUVS limb scanning
 2. STATIC measurement of ions escaping upwards
 3. NGIMS to scoop up thermal species in s/c ram direction



SUMMARY

- MAVEN's orbital period, inclination, and periapsis altitude will provide the best comprehensive coverage of Mars escape-related regions possible for a one Earth-year mission
- The instruments, which have high heritage, will sample all escape processes
- Phasing of the mission on the declining phase of the solar cycle maximizes the range of solar variability inputs needed for extrapolating loss vs. solar inputs backwards in the history of the solar system
- In addition theoretical modeling efforts are planned to provide closure in understanding