

# SIMBA

## Measuring the Earth's Radiation (Im)balance

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Absolute Radiometry

Royal Meteorological Institute of Belgium (RMIB)

Solar-Terrestrial Centre of Excellence (STCE)

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## SIMBA

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### Context

Absolute Radiometry

### SIMBA Mission

Missions Objectives

Design

Challenges

### Beyond SIMBA

Mission's expectations

Our future with cubesats

## Context — who are we ?

Absolute radiometry : a brief introduction

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# RMIB, Laboratory for Absolute Radiometry

Who are we?

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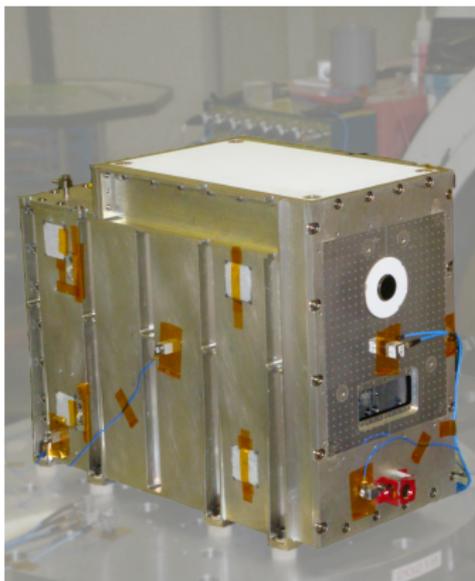
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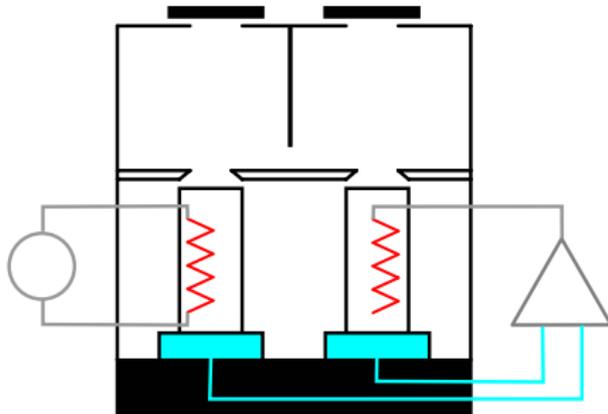
SovaP radiometer during  
PICARD integration



Pyrheliometer Comparison in  
Davos, CH

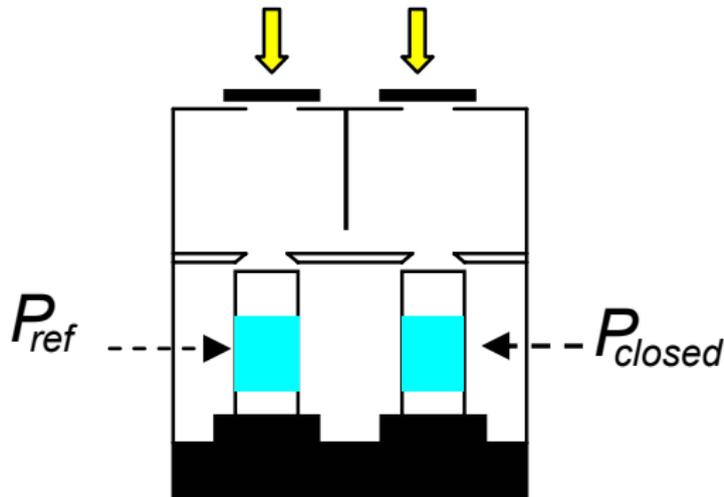
## Schematic principle of a Differential Absolute Radiometer (DIARAD)

### Cavities, Heaters and Flux Sensors



## Schematic principle of a Differential Absolute Radiometer (DIARAD)

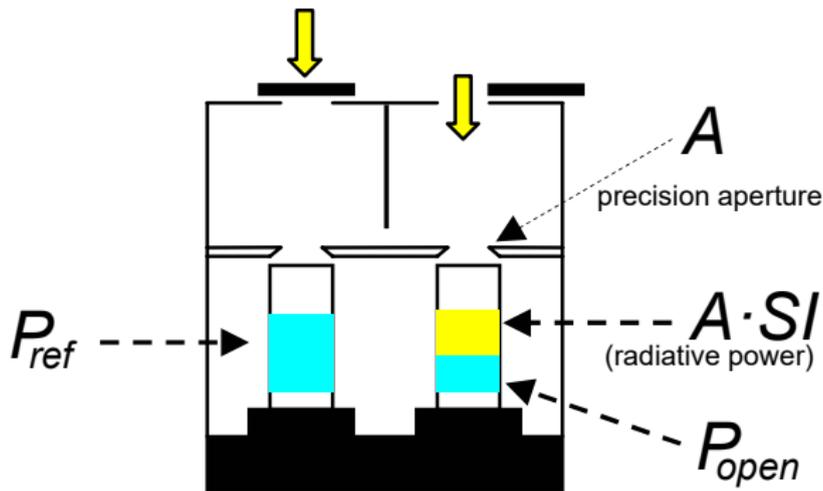
### CLOSED PHASE (90 seconds)



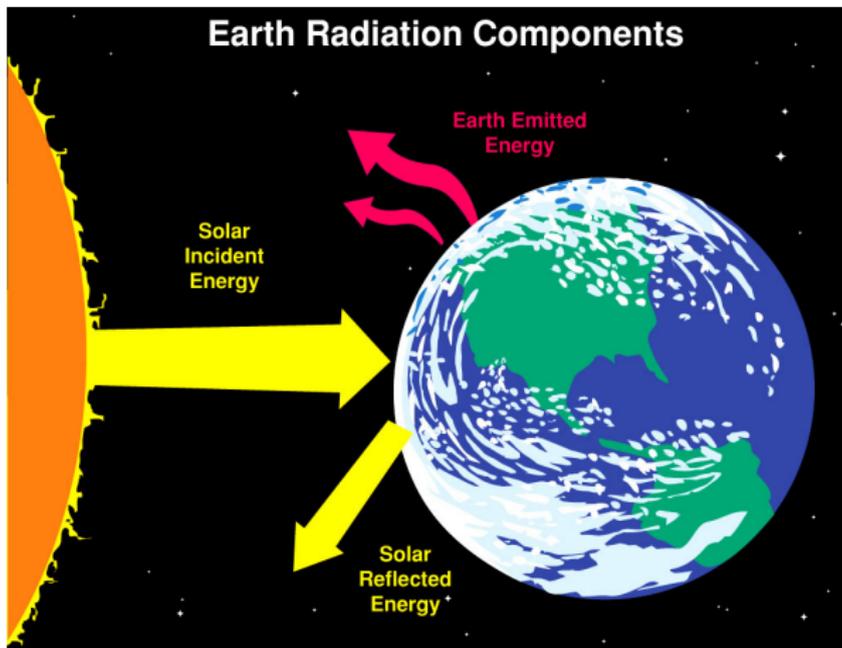
Both shutters are closed

## Schematic principle of a Differential Absolute Radiometer (DIARAD)

OPEN PHASE (90 seconds)



Active shutter open



# Partners of the Mission

The goals of the SIMBA mission

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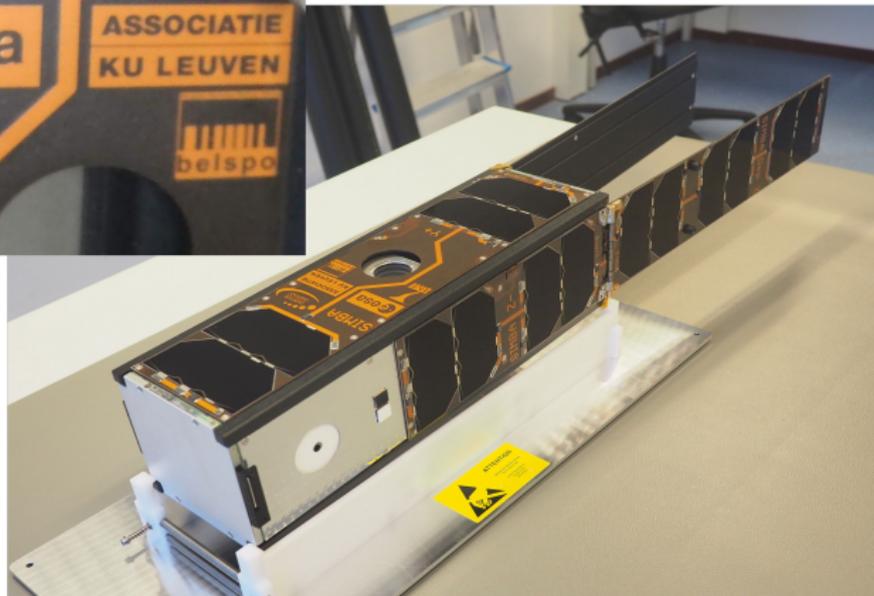
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# Satellite and Bus

## Design

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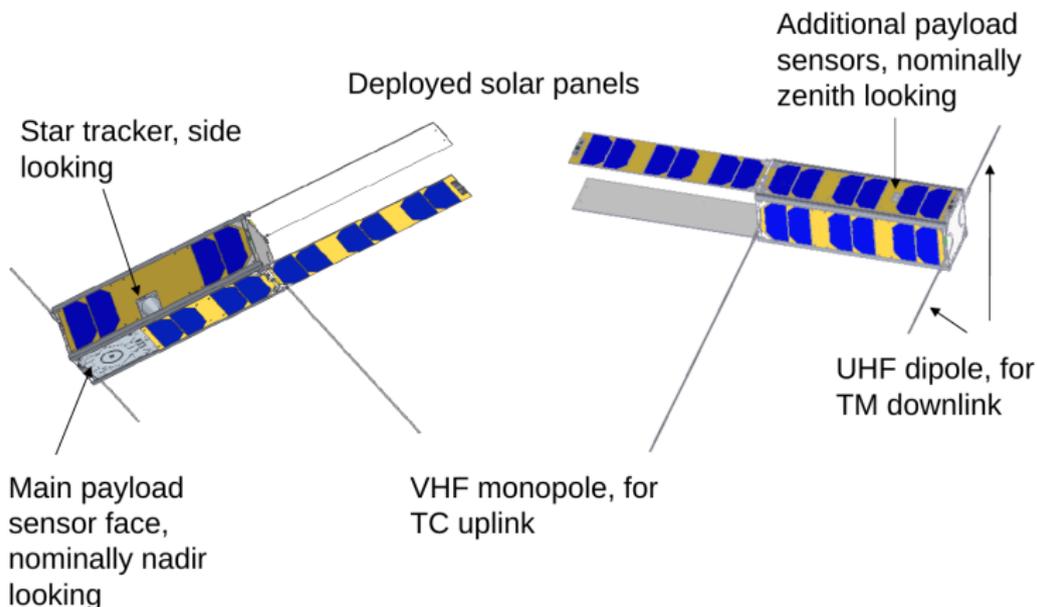
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▶ 3U, 4 kg, 3.5 W, 0.6 MB/day

▶ Payload and lead: RMIB, ADCS: KUL, Bus: ISIS

# Cavity sensors

Payload bay and sensors array

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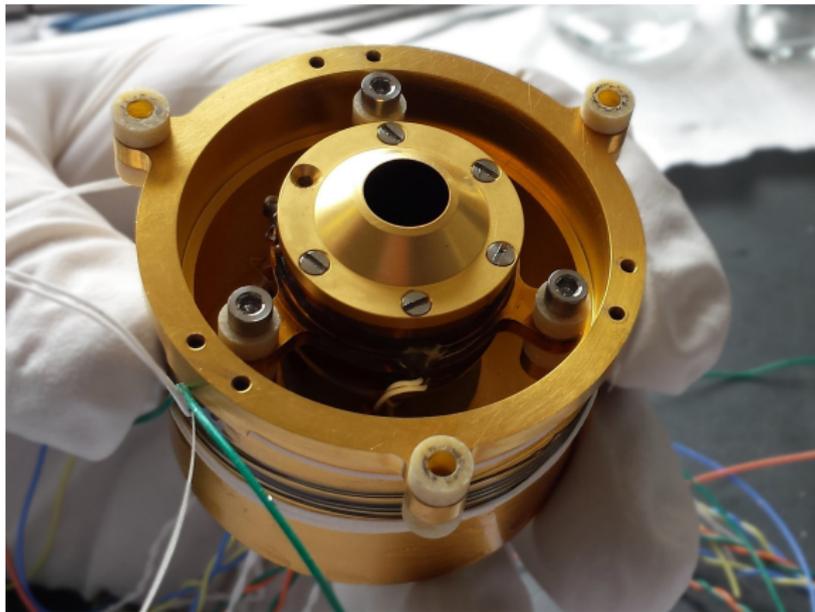
**Design**

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# Flat Spectrum Sensors

Payload bay and sensors array

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# Payload bay integration

Payload bay and sensors array

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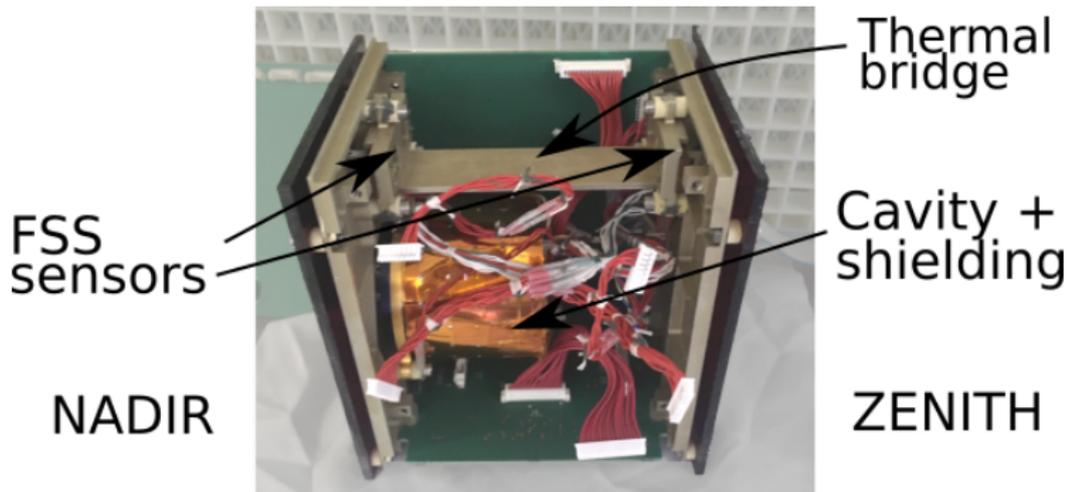
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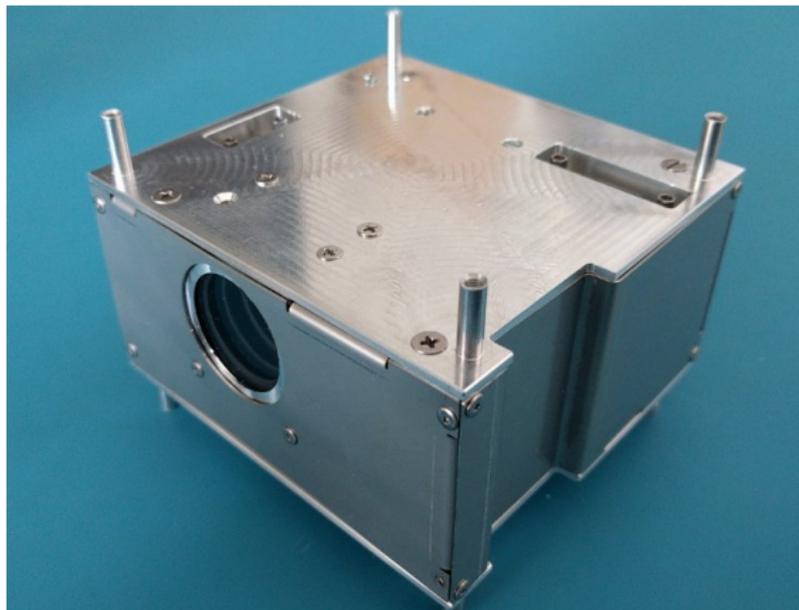
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Star tracker + inertial gyroscopes.  $\delta \triangleleft \approx 0.1$

# Mission's current status

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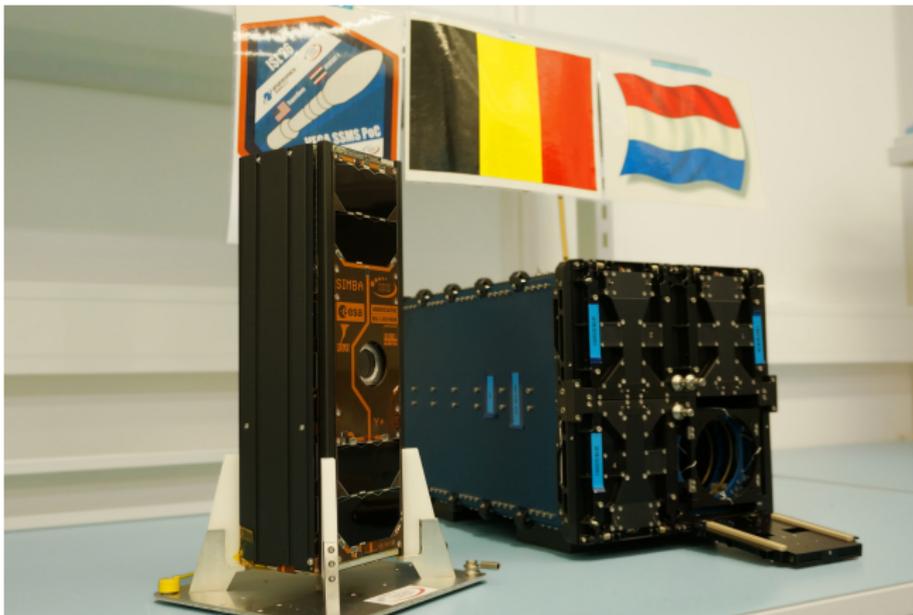
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Ready for vehicle integration, shall be launched by Vega VV-16 in March 2020 (Delayed after Vega VV-15 failure)

Enormous overhead compared to just piggyback on a (big) satellite

- ▶ Payload is main contractor for the whole (albeit small) satellite
- ▶ Think about legal aspects to operate
- ▶ Cubesats limitations
- ▶ Micro electronic integration
- ▶ Radio frequencies (for science missions the amateur option is no longer an option)
- ▶ ...

## Primary objective

- ▶ Measure Essential Climate Variables :
  - ▶ Total Solar Irradiance
  - ▶ Earth Radiation Budget
- ▶ Characterize the instrument's accuracy and limitations

## A future constellation ?

A larger fleet would be able integrate the measures worldwide and monitor the Sun-Earth Imbalance

## Examine a 12U solar radiometer cubesat

- ▶ Can fit a conventional DIARAD radiometer
- ▶ Has enough power
- ▶ Sufficient mass for thermal insulation
- ▶ Longer lifetime (higher mass and redundant systems)

## Test new and examine old technology in orbit

- ▶ Paint or nanotubes
- ▶ Cryogenic radiometers
- ▶ Radiometer topologies
- ▶ Electronics

# Thank you for your attention!

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